

Cryogenic

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**Gravitational Wave Challenges and Cosmology
@International Institute of Physics , Natal, Brazil**

Abstract

Why do we adopt cryogenics (**cooled mirrors**)
to detect gravitational wave ?

(1) Merits : Reduction of **thermal noise**
and **so on**

(2) How to **cool down**

Contents

- 1. Introduction***
- 2. Thermal noise***
- 3. Other merits***
- 4. How to cool down***
- 5. Summary***

1. Introduction

List of **cryogenic** interferometric gravitational wave detectors

Constructed

CLIO (Japan, **Sapphire**, 100m)

KAGRA (Japan, **Sapphire**, 3km)

Future plan

Voyager (U.S.A., **Silicon**, 4km, LIGO facility,
Odylio Aguiar in this afternoon)

Einstein Telescope (Europe, **Silicon**, 10km)

Cosmic Explorer (U.S.A., **Silicon**, 40km)

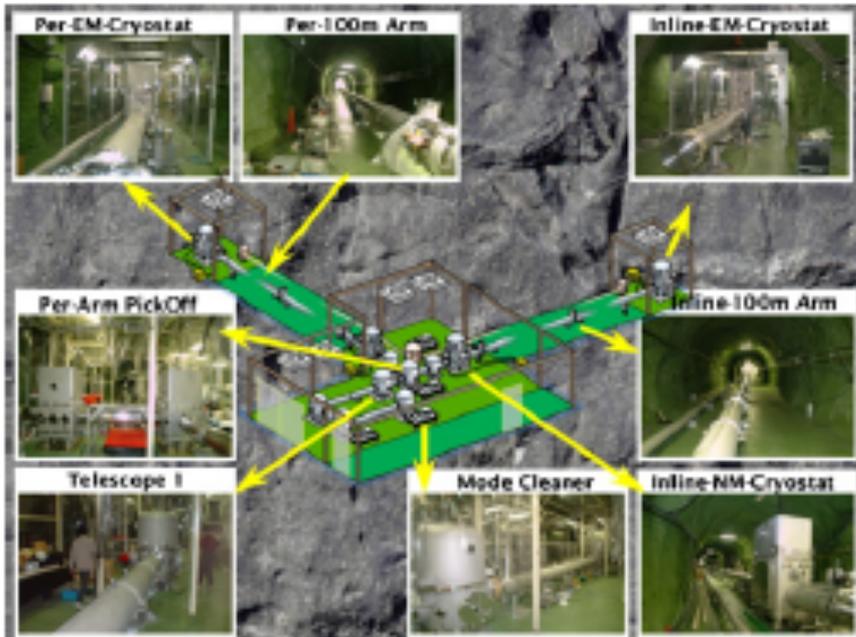
(Michele Punturo in this afternoon)

1. Introduction

List of **cryogenic** interferometric gravitational wave detectors

Constructed

CLIO (Japan, Sapphire, 100m)



K.Agatsuma et al., Physical Review Letters 104 (2010) 040602.

T. Uchiyama et al., Physical Review Letters 108 (2012) 141101.

1. Introduction

List of **cryogenic** interferometric gravitational wave



KAGRA sapphire suspension

(**100m**)

(**3km**, observation starts

(**4km**, LIGO facility)

(**10km**)

(**40km**)

1. Introduction

List of **cryogenic** interferometric gravitational wave



KAGRA sapphire suspension

There are 4 mirrors.
3 of them are cooled down. The last one will be cooled down within two weeks.

on, 4km, LIGO facility)

Europe, **Silicon**, 10km)

.A., **Silicon**, 40km)

1. Introduction

List of **cryogenic** interferometric gravitational wave



KAGRA sapphire suspension

Commissioning with cooled mirror is in progress.

Engineering run of a 3km cavity on 8th of June 2019. Duty cycle is 95% at least.

(A., SILICON, 40KHz)

1. Introduction

List of **cryogenic** interferometric gravitational wave



KAGRA sapphire suspension

Fabian Arellano, Enzo Tapia, Koki Okutomi will present KAGRA vibration isolation system tomorrow afternoon.

on, 4km, LIGO facility)

Europe, Silicon, 10km)

.A., Silicon, 40km)

2. *Thermal noise*

Motivation of cryonic interferometer is **reduction of thermal noise**.

Story is not so simple ...

Energy equipartition theorem

$$\langle x^2 \rangle = k_B T/2$$

We are interested with **power spectrum density in frequency region**. Our observation band is between **10Hz and 10kHz**. x^2 is integral of power spectrum density **in all frequency region**.

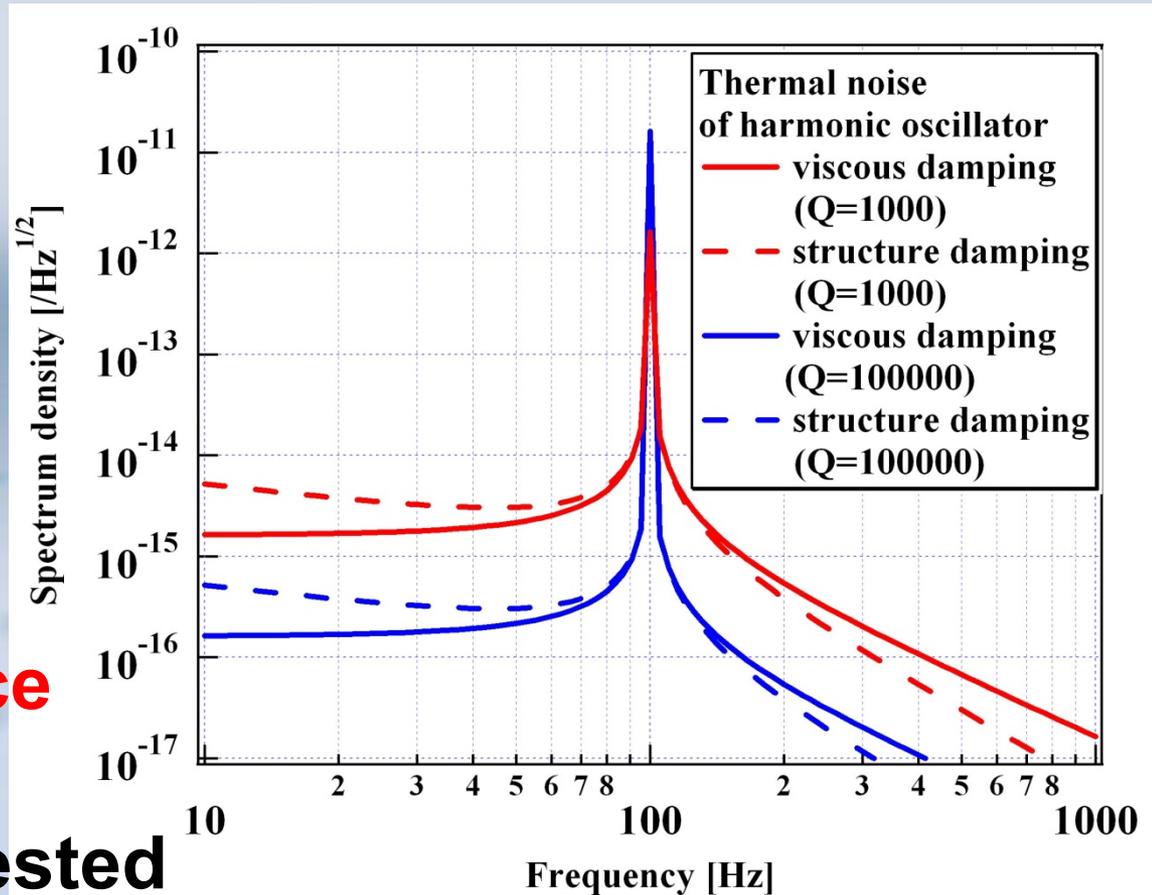
2. Thermal noise

Spectrum density of **thermal noise** of **harmonic oscillator**

Q-value :
Magnitude of loss

Higher Q is smaller loss.

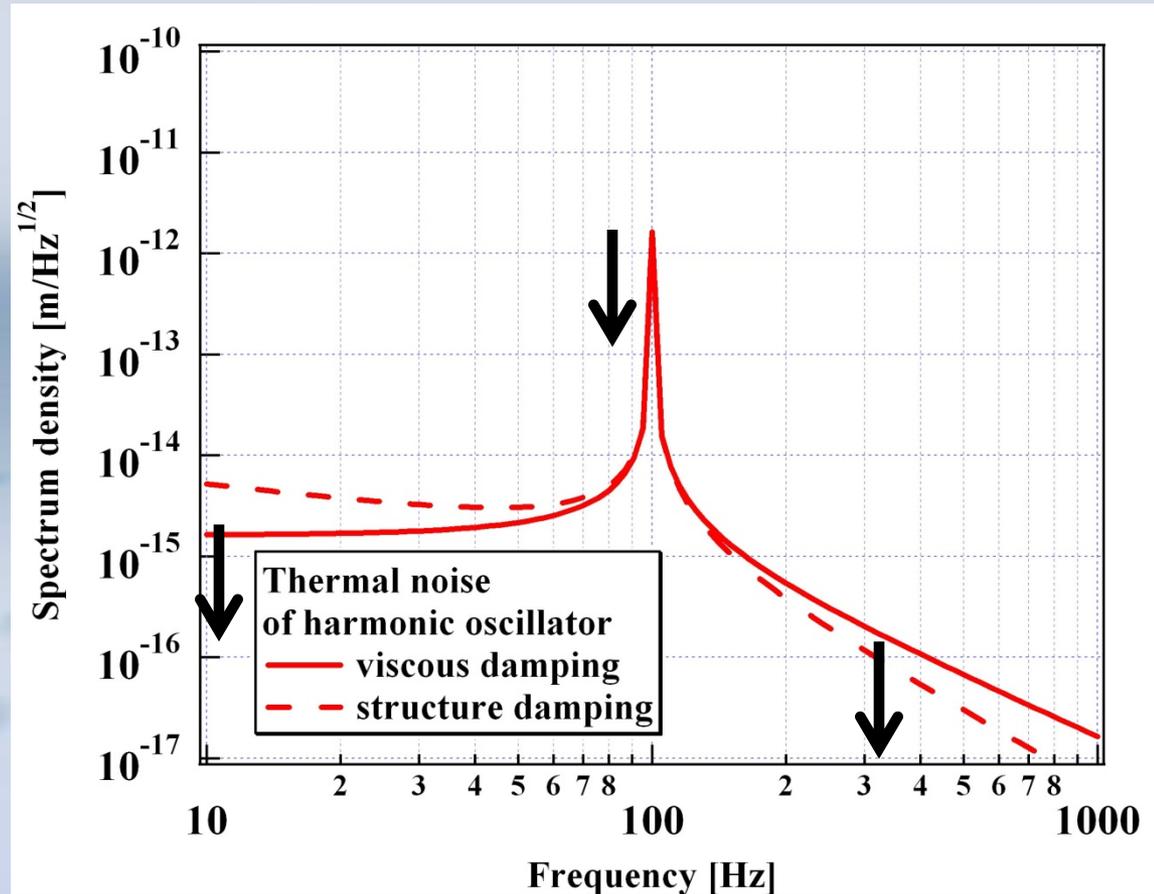
Higher Q is smaller off resonance thermal noise and better. We interested with off resonance.



2. Thermal noise

Spectrum density of **thermal noise**
of **harmonic oscillator**

When Q is independent of temperature, power spectrum density **in all frequency region** is **smaller** at **lower temperature**.



2. *Thermal noise*

Amplitude of thermal noise (off resonance) is proportional to

$$(T/Q)^{1/2}$$

In general, **Q-value depends on T** (temperature).

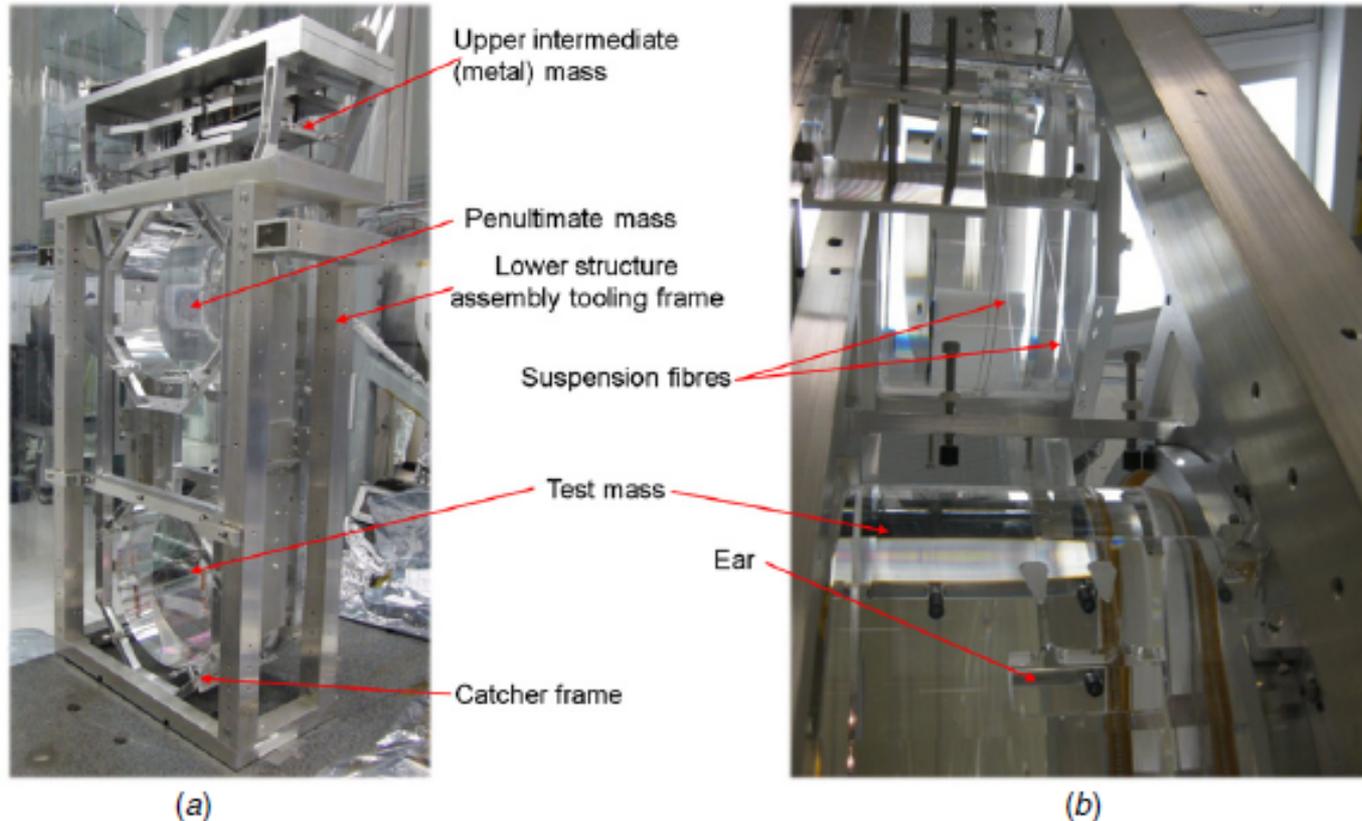
We must investigate how dissipation depends on temperature in **cryogenic region**.

2. Thermal noise

Room temperature second generation interferometer
Fused silica mirror suspended by **fused silica** fibers

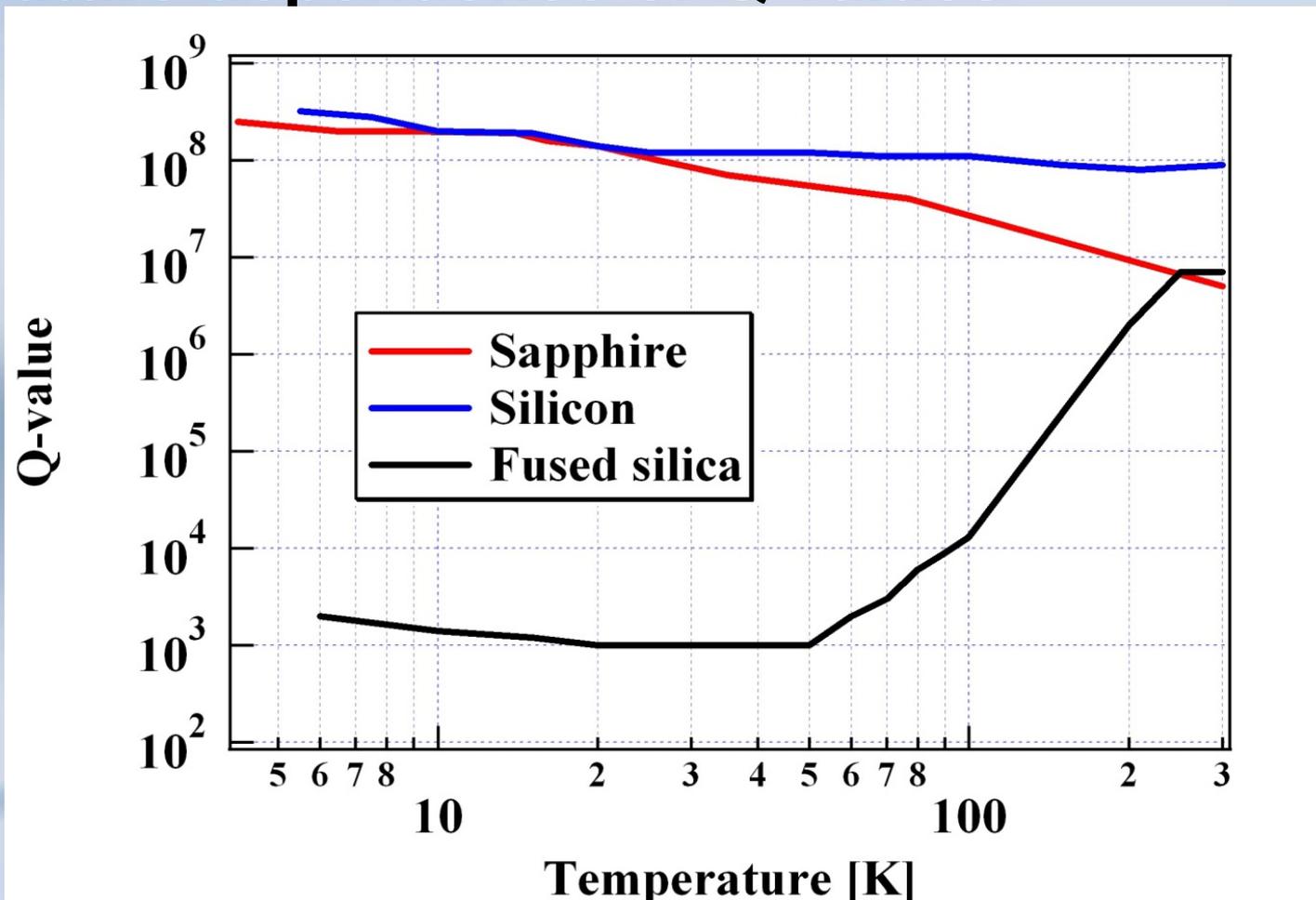
Class. Quantum Grav. 29 (2012) 035003

A V Cumming *et al*



2. Thermal noise

Temperature dependence of Q values



T. Uchiyama *et al.*, Physics Letters A 261 (1999) 5-11.

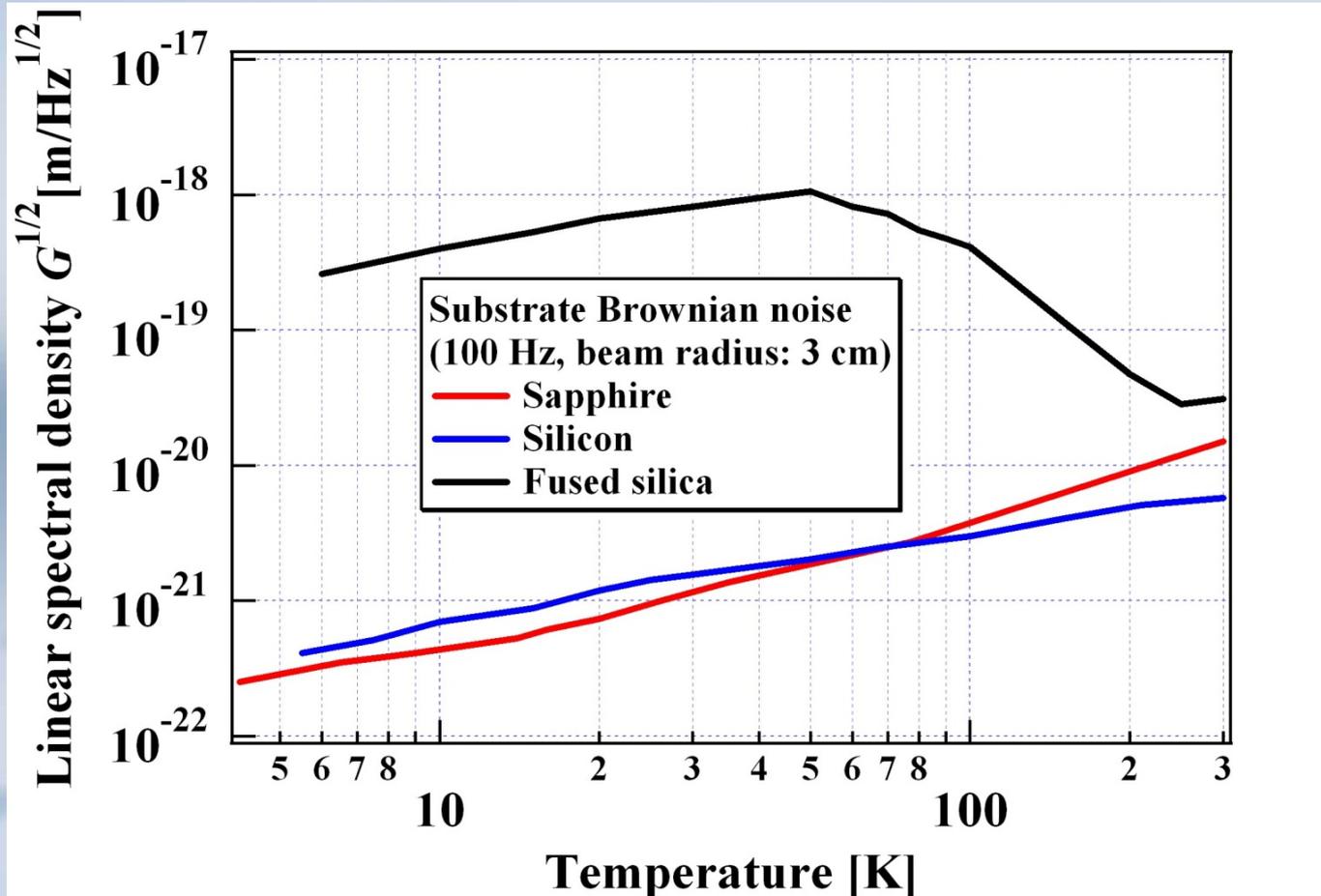
R. Nawrodt *et al.*, Journal of Physics: Conference Series 122 (2008) 012008.

C. Schwarz *et al.*, 2009 Proceedings of ICEC22-ICMC2008.

2. Thermal noise

Temperature dependence of $(T/Q)^{1/2}$

$(T/Q)^{1/2}$



T. Uchiyama *et al.*, Physics Letters A 261 (1999) 5-11.

R. Nawrodt *et al.*, Journal of Physics: Conference Series 122 (2008) 012008.

C. Schwarz *et al.*, 2009 Proceedings of ICEC22-ICMC2008.

2. *Thermal noise*

Important message

We can not **use fused silica**
for **cryogenic** interferometer.

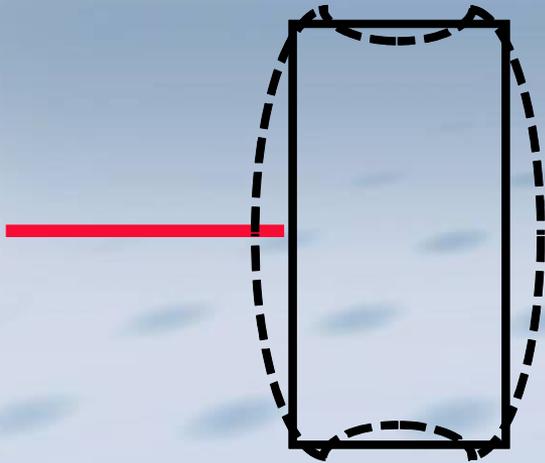
New material (Sapphire, Silicon) is necessary for
mirror substrate and fibers to suspend mirrors.

Challenge !

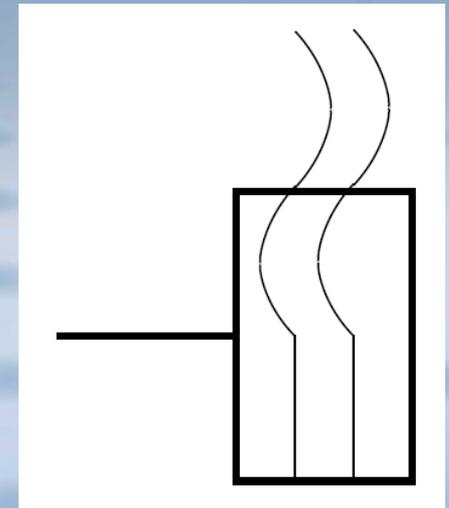
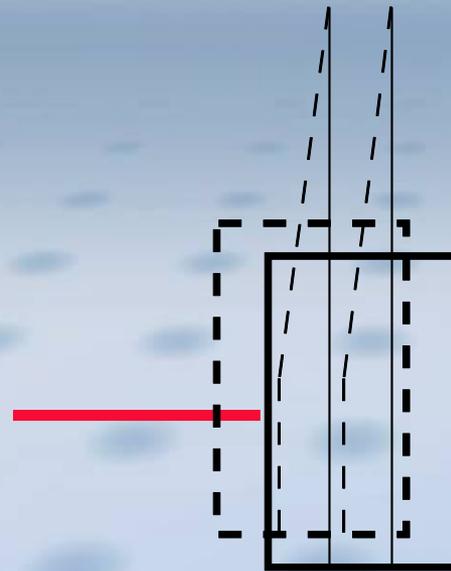
2. Thermal noise

Thermal noise of **mirror** and **suspension**

Mirror



Suspension

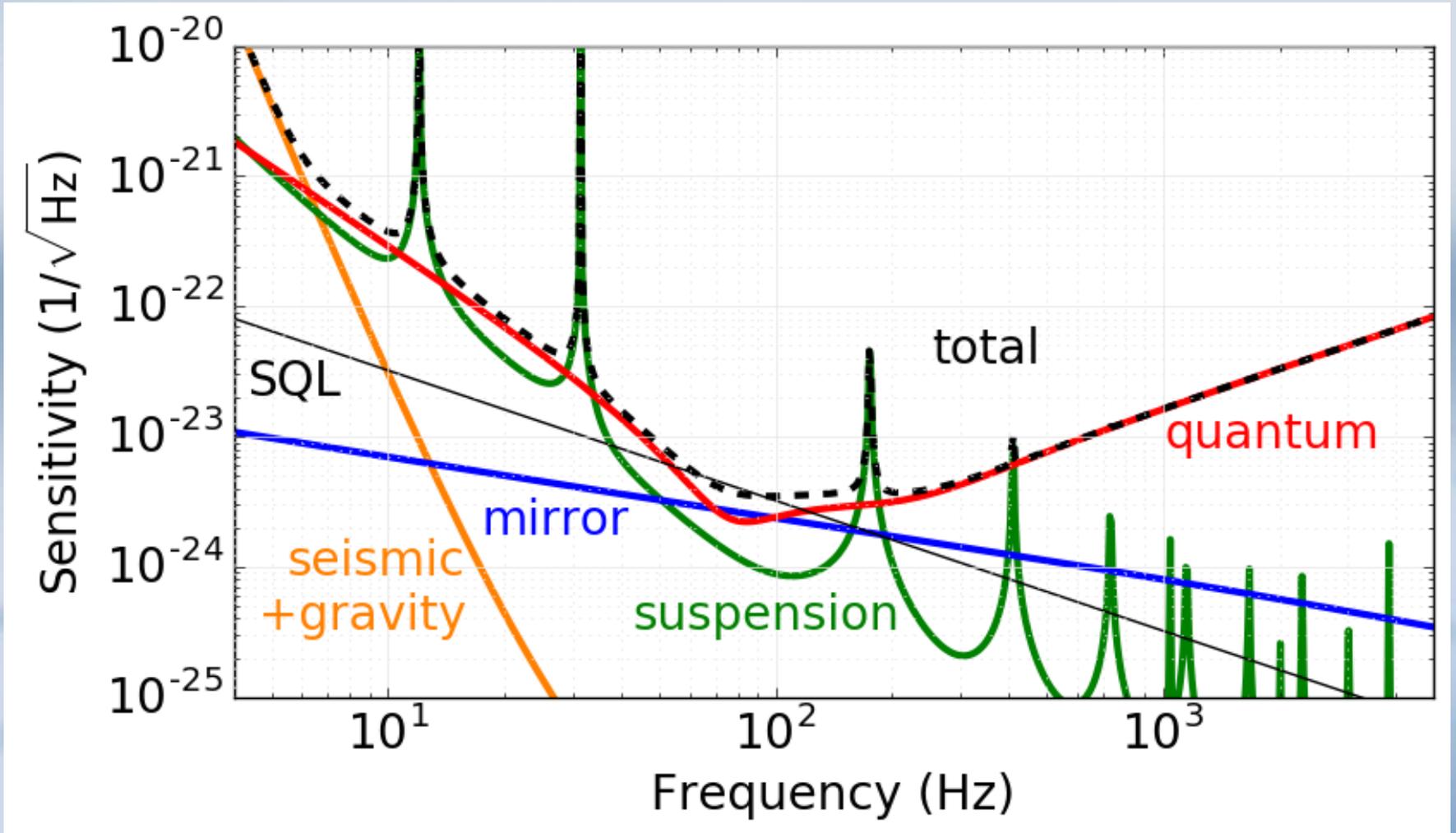


Pendulum

Violin modes

2. Thermal noise

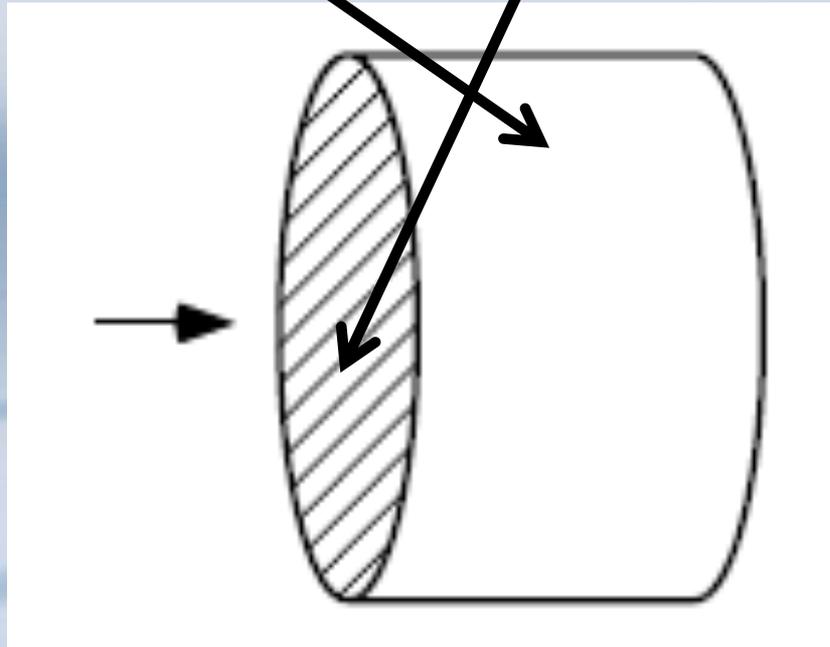
Thermal noise of **mirror** and **suspension**



2. *Thermal noise*

Thermal noise of **mirror**

Not only substrate but also coating must be taken into account !



Y. Levin, *Physical Review D* 57 (1998) 659.

2. *Thermal noise*

Mechanical dissipation generates thermal noise.

Loss in **substrate** and **coating**

Two kinds of loss : **Structure**, **Thermoelastic**

Structure : Unknown frequency independent .

Measurement of Q-values is only reliable way.

P.R. Saulson, Phys. Rev. D 42 (1990) 2437.

Thermoelastic : Coupling of temperature fluctuation and thermal expansion coefficient. We can calculate.

C. Zener, Phys. Rev. 52 (1937) 230; *ibid.* 53 (1938) 90.

2. *Thermal noise*

Mechanical dissipation generates thermal noise.

Loss in *substrate* and *coating*

Two kinds of loss : *Structure*, *Thermoelastic*

So, there are four kinds of loss

(mirror thermal noise).

(1) Structure damping in substrate

(2) Thermoelastic damping in substrate

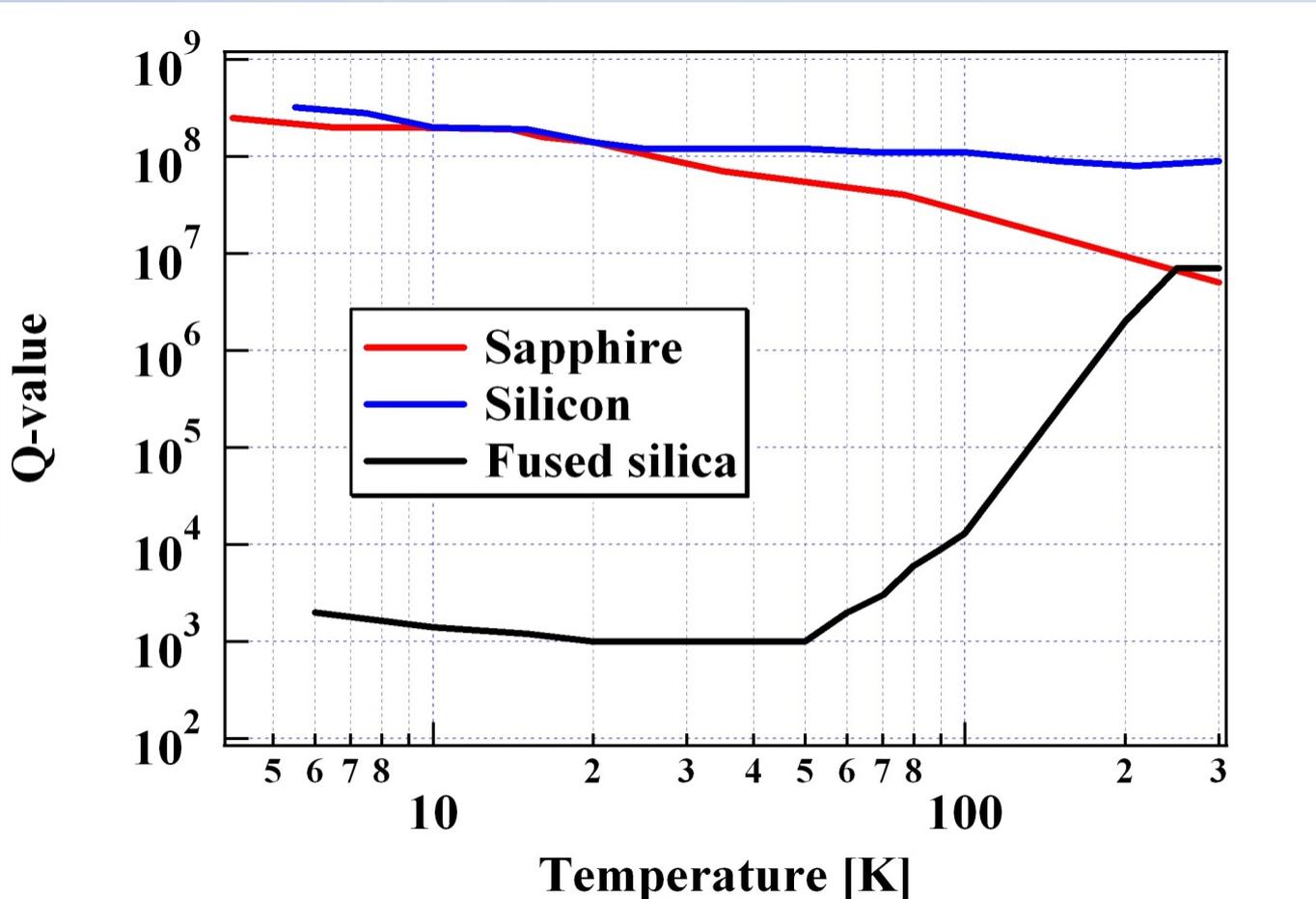
(3) Structure damping in coating

(4) Thermoelastic damping in coating

2. Thermal noise

Structure damping in substrate

Inverse number of loss strength



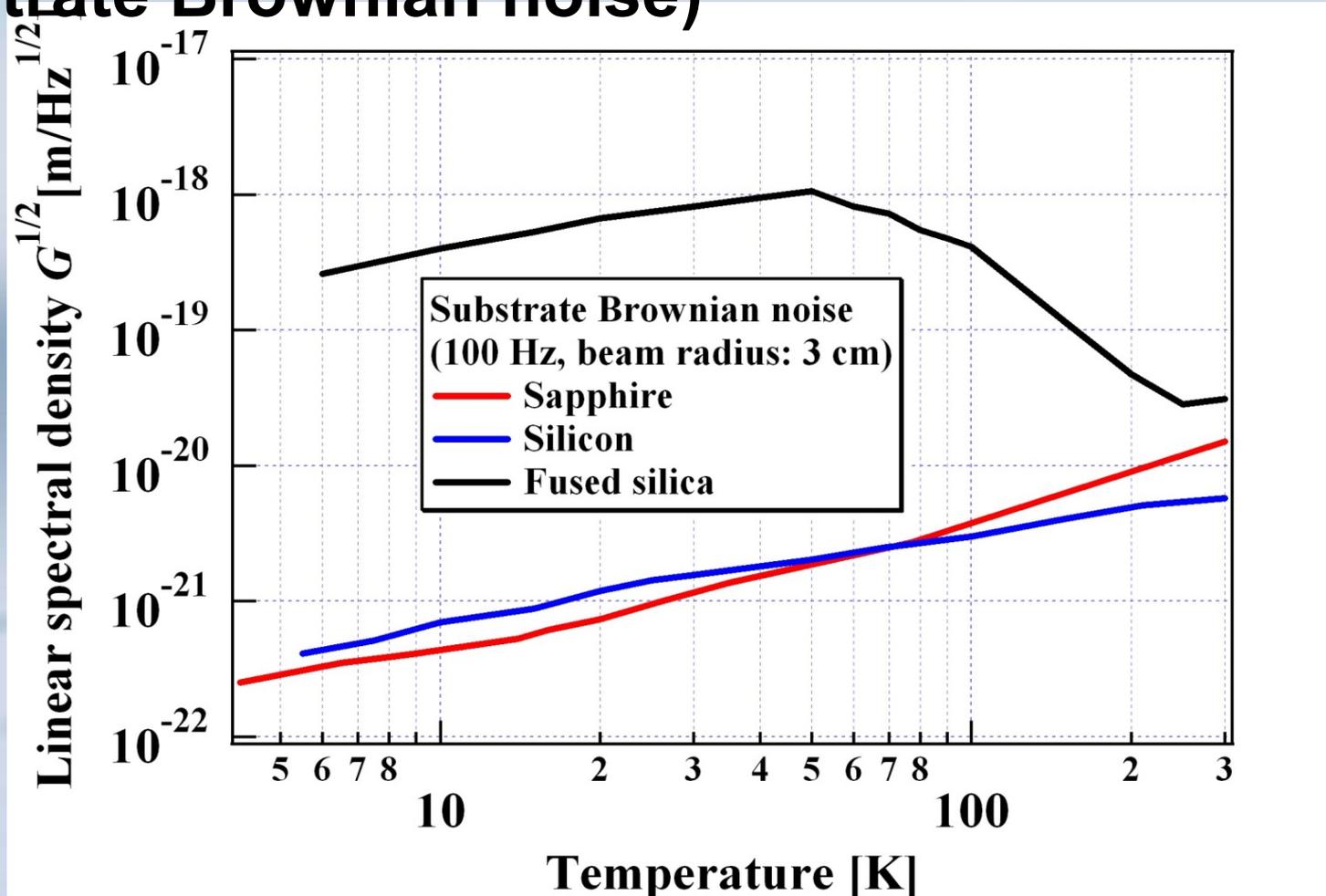
T. Uchiyama *et al.*, Physics Letters A 261 (1999) 5-11.

R. Nawrodt *et al.*, Journal of Physics: Conference Series 122 (2008) 012008.

C. Schwarz *et al.*, 2009 Proceedings of ICEC22-ICMC2008.

2. Thermal noise

Thermal noise by structure damping in substrate
(Substrate Brownian noise)

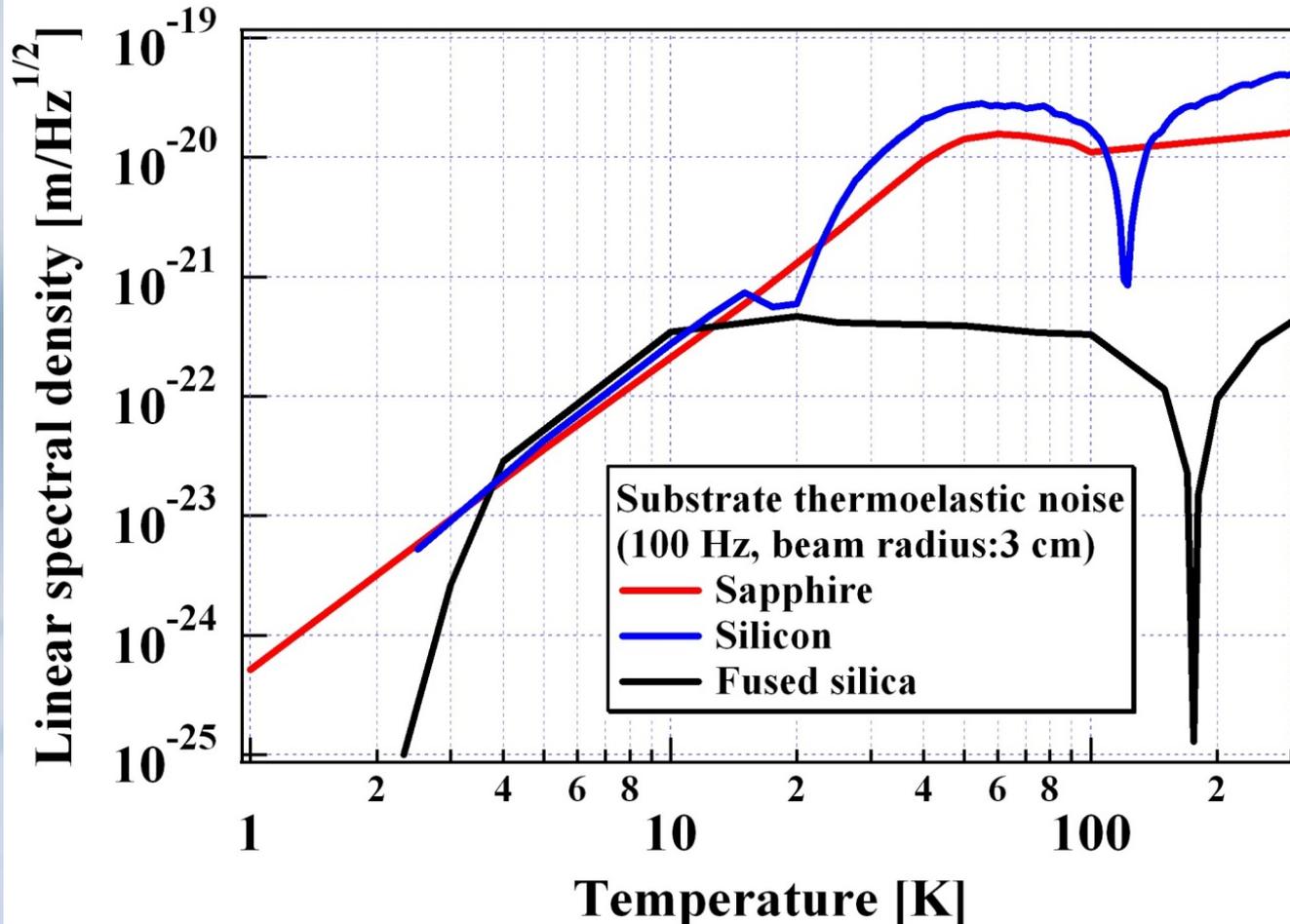


2. Thermal noise

Thermal noise by thermoelastic damping

M. Cerdonio *et al.*, Phys. Rev. D 63 (2001) 082003.

in substrate

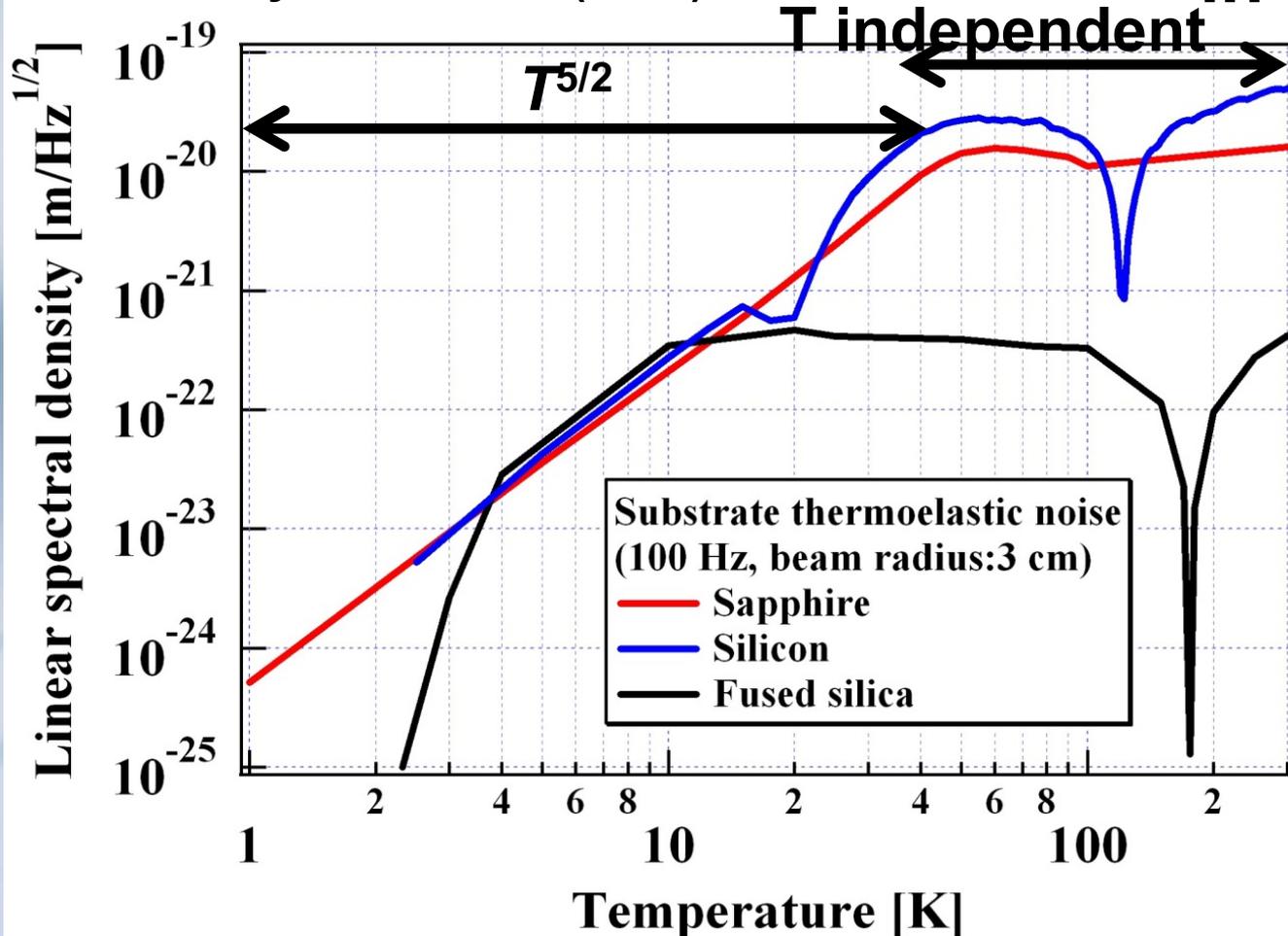


2. Thermal noise

Thermal noise by thermoelastic damping

in substrate

M. Cerdonio *et al.*, Phys. Rev. D 63 (2001) 082003.

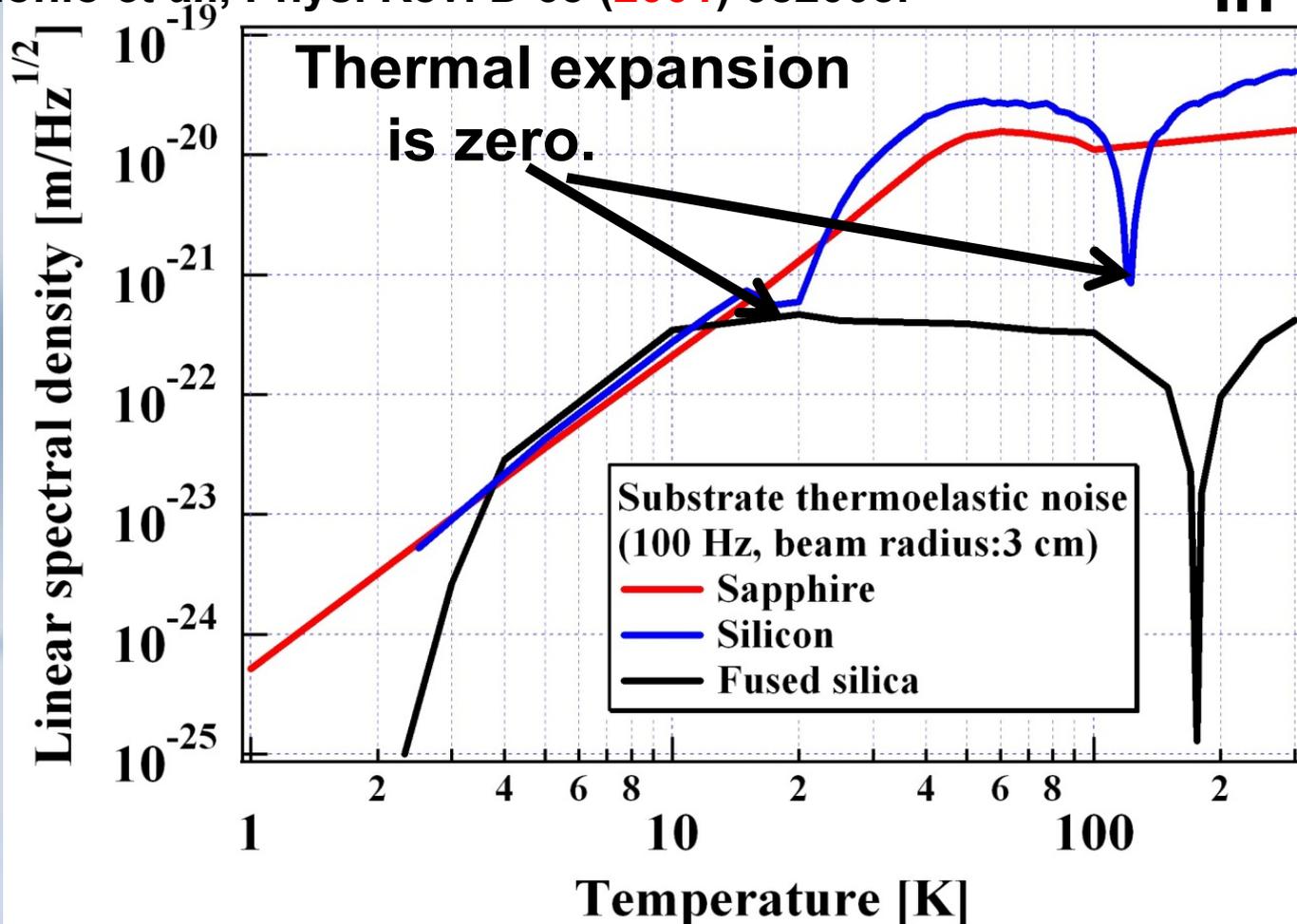


2. Thermal noise

Thermal noise by thermoelastic damping

in substrate

M. Cerdonio *et al.*, Phys. Rev. D 63 (2001) 082003.



2. *Thermal noise*

Structure damping in **coating**

Y. Levin, Phys. Rev. D 57 (**1998**) 659.

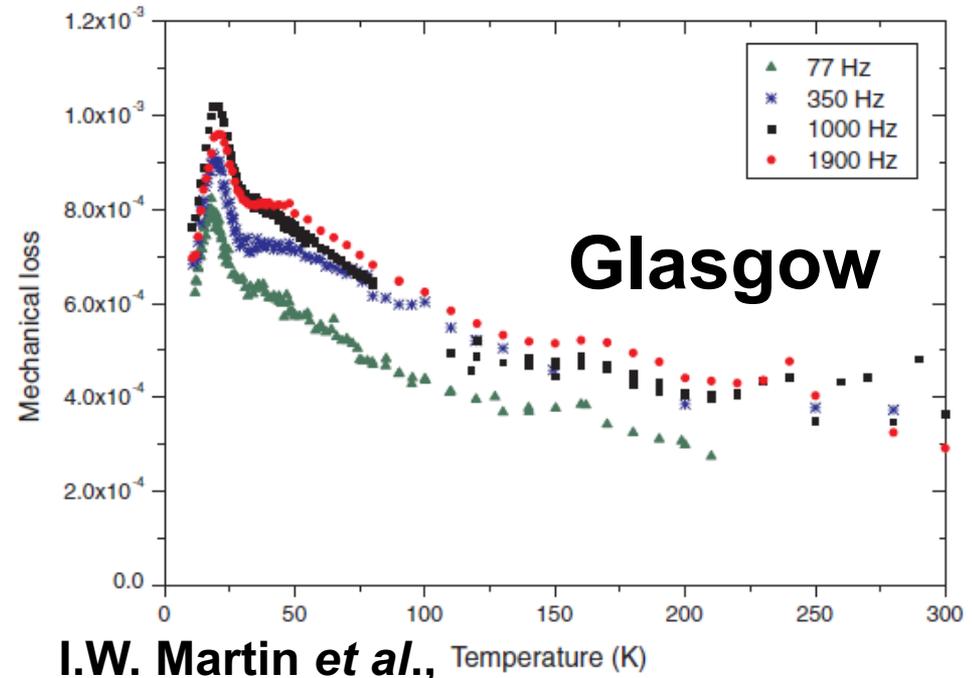
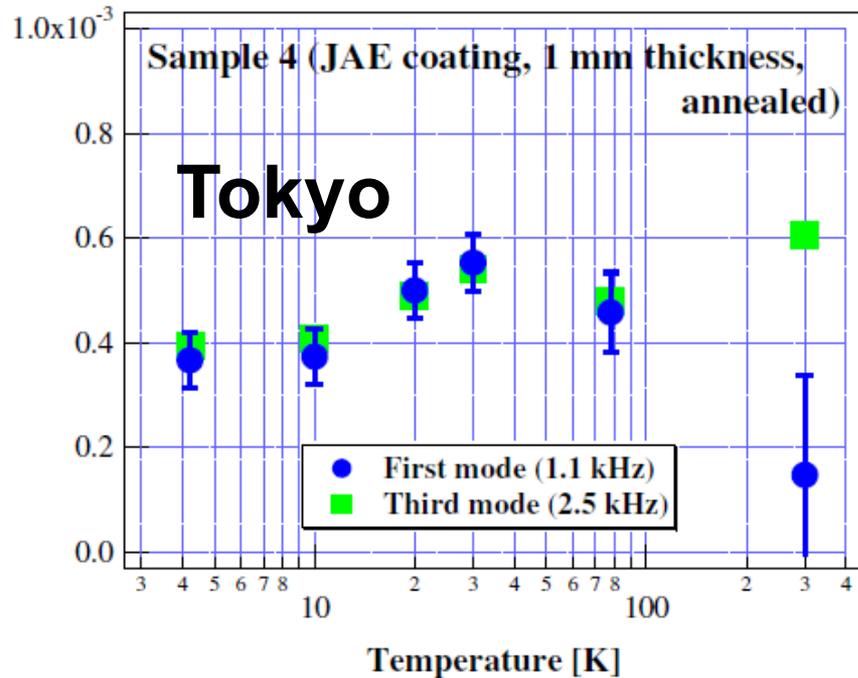
Coating has **large contribution** to thermal noise
than we expected !

This noise is proportional to $\phi^{1/2}$.

ϕ is **loss strength** (it is inverse number of Q-values).

2. Thermal noise

Structure damping in coating (Ta_2O_5/SiO_2)



K. Yamamoto *et al.*,
Physical Review D 74 (2006) 022002.

I.W. Martin *et al.*, Temperature (K)

Classical and Quantum Gravity
27 (2010) 225020.

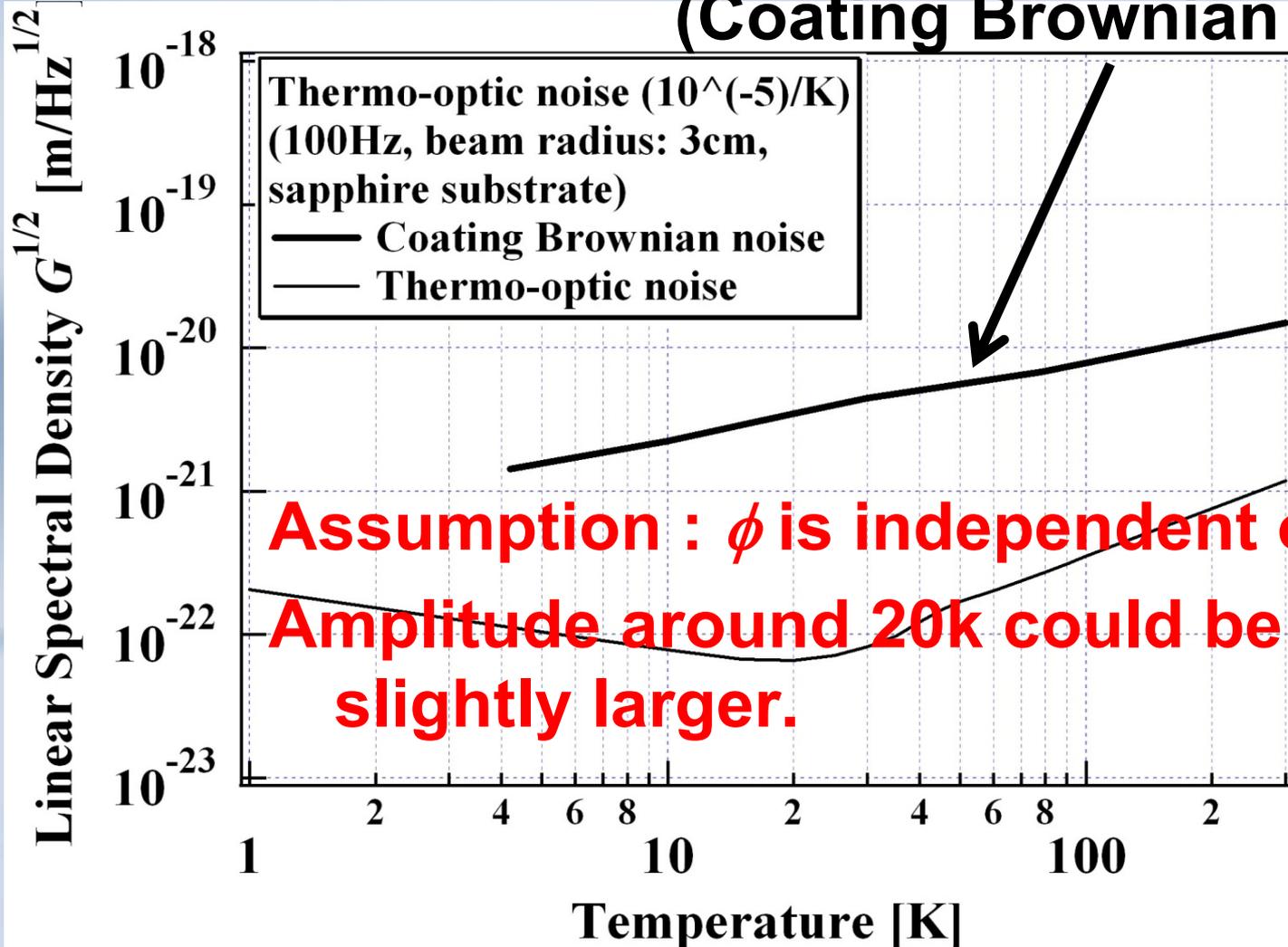
(Annealing could suppress peak)

Many experiments show peak around 20K.

2. Thermal noise

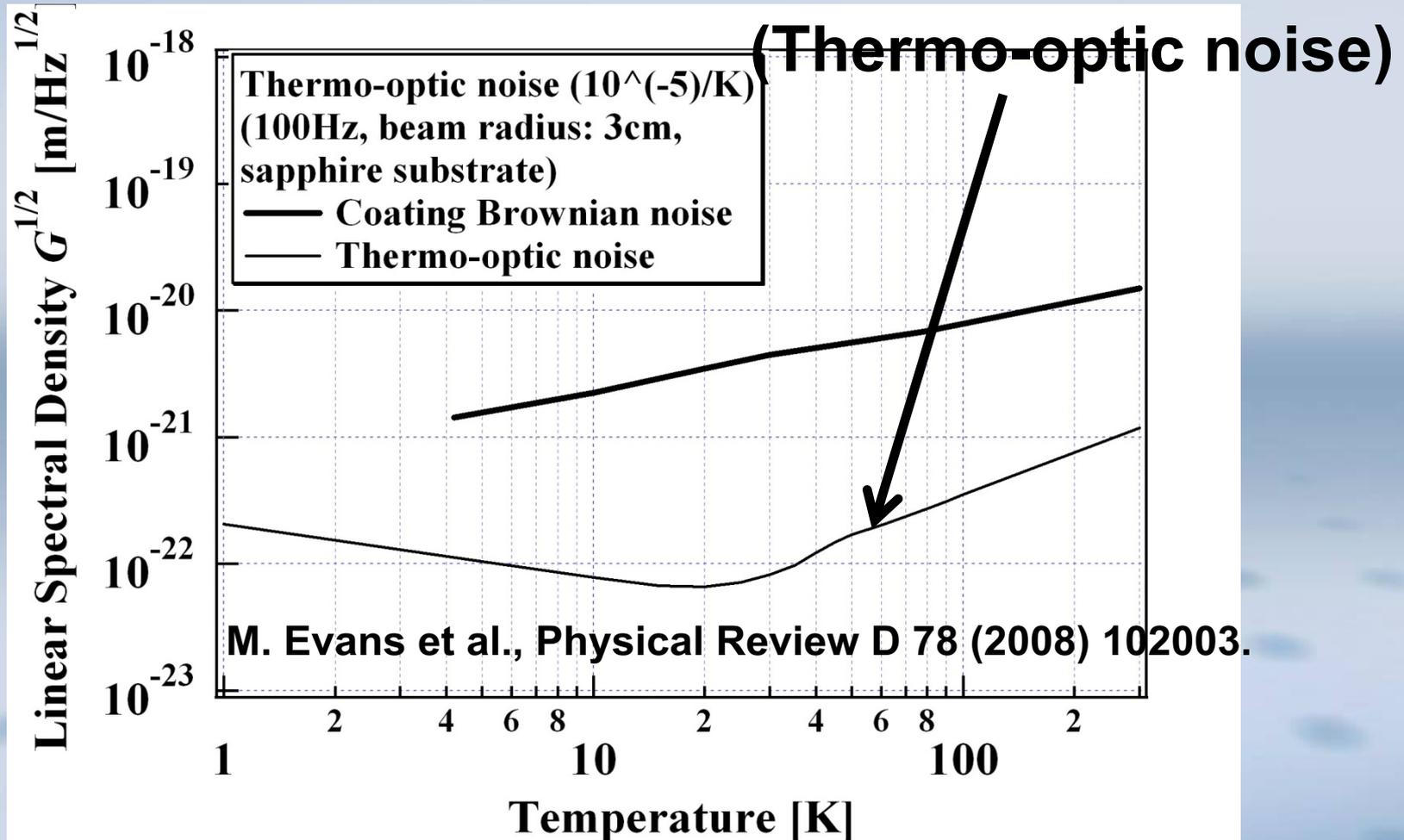
Thermal noise by structure damping in **coating**

(Coating Brownian noise)



2. Thermal noise

Thermal noise by thermoelastic damping in **coating**



Thermo-optic noise is much smaller than coating Brownian noise.

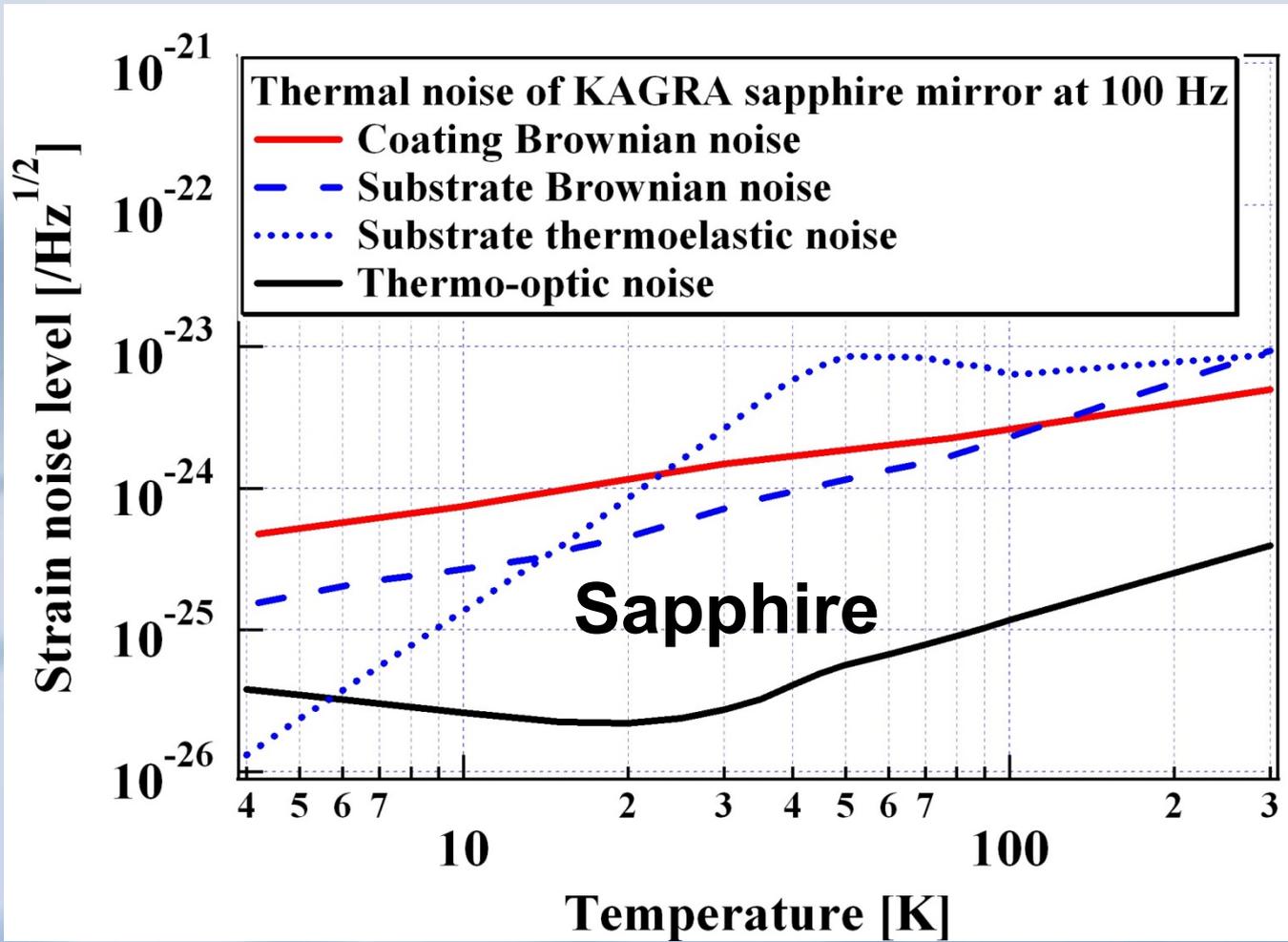
2. Thermal noise

Summary : **Sapphire** mirror

In principle, **lower temperature is better.**

> 50K
Constant

<20K
Enough
small



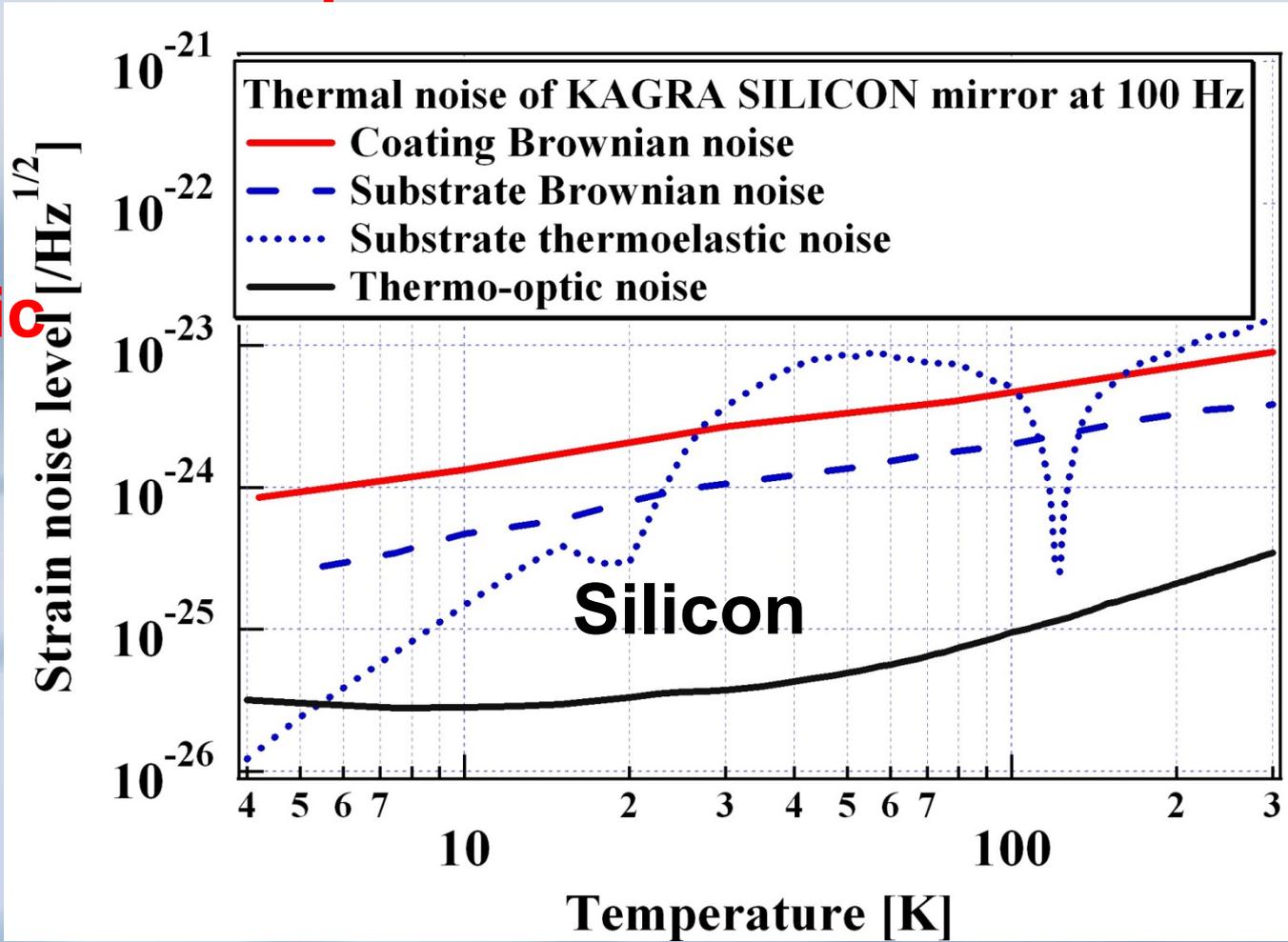
2. Thermal noise

Summary : **Silicon** mirror

In principle, **lower temperature is better.**

120K
Thermoelastic noise
vanishes.

Temperature control is necessary.



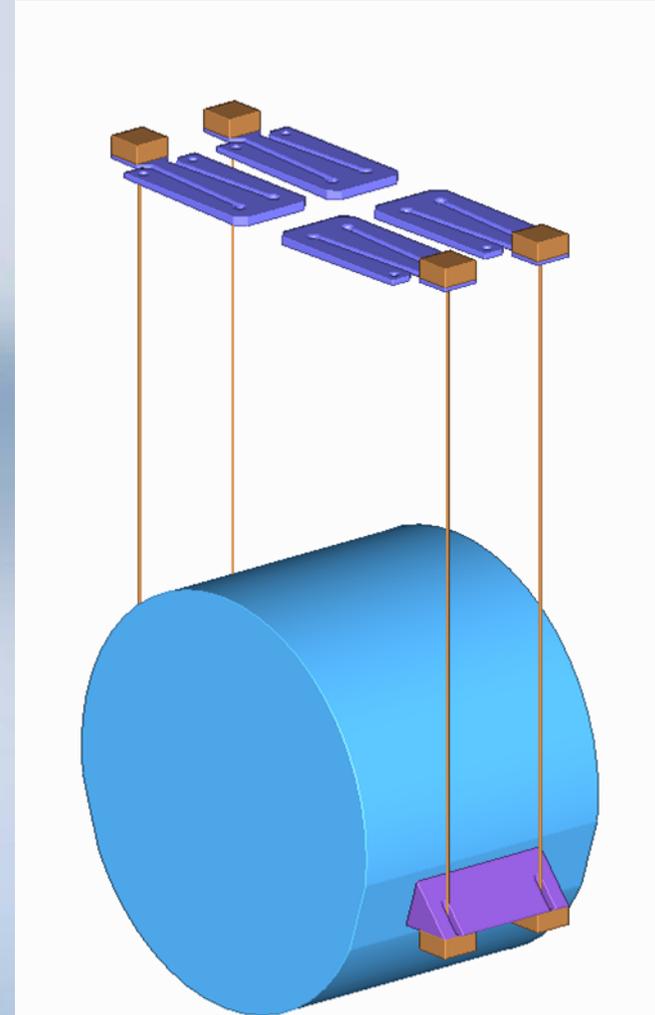
2. *Thermal noise*

Mirror substrate and **operation temperature**

Sapphire, 20 K : CLIO, KAGRA

Silicon, 10 K : Einstein Telescope

Silicon, 120K : Voyager
Cosmic Explorer



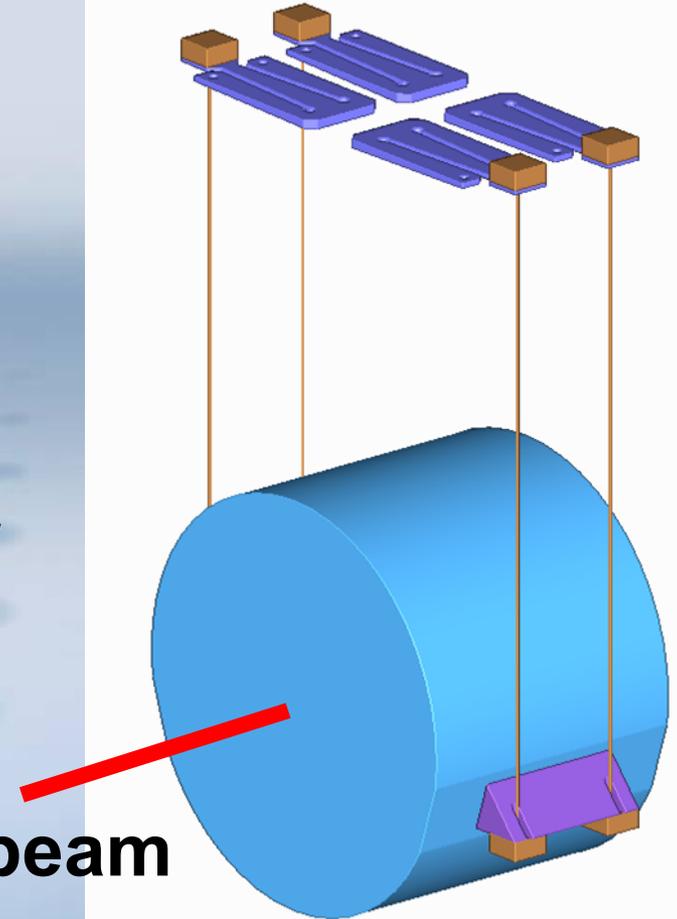
2. *Thermal noise*

Mirror substrate and **operation temperature**

Heat **absorption** in mirror is an serious issue. How can we avoid that mirror could be **hot** ?

Typical heat absorption in mirror
1W – 10 W

Laser beam



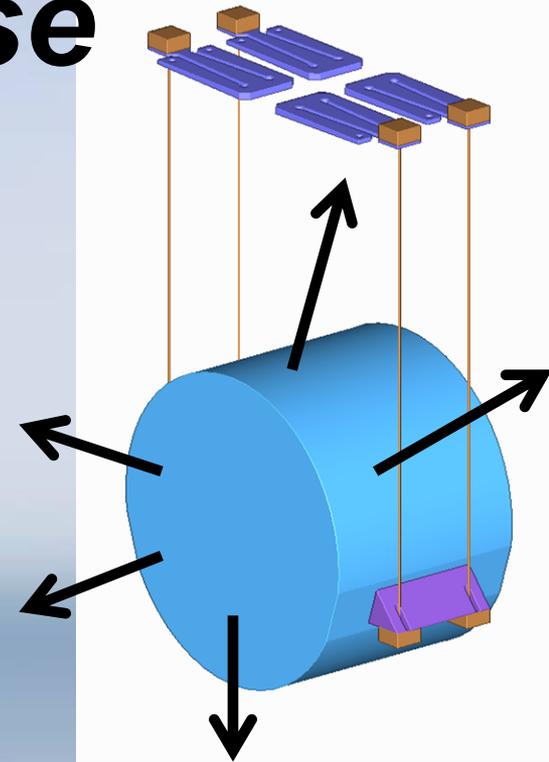
2. *Thermal noise*

120K operation (Silicon)

Heat absorbed in mirror : **Several W**

Radiation from mirror is effective.

When mirror is surrounded by **cold wall** (radiation shield), radiation from mirror carries much heat ; **13 W/m²** at most (Black body radiation). Surface of mirror is 0.3 m² at least.



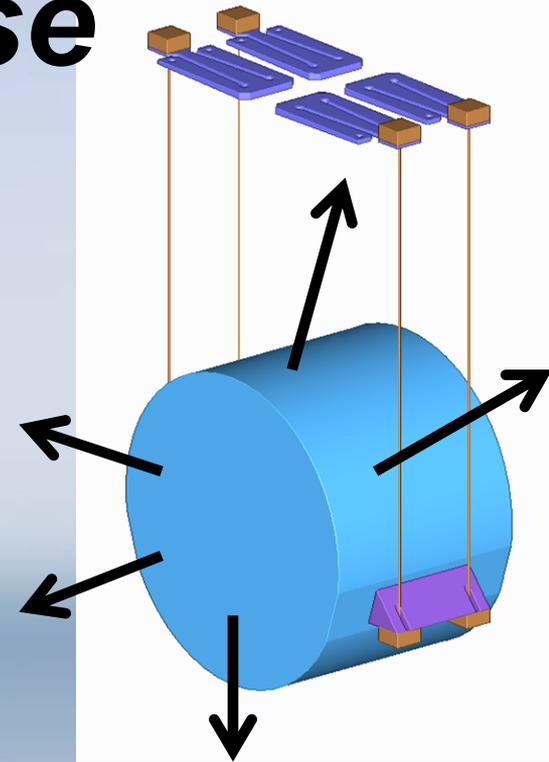
2. *Thermal noise*

120K operation (Silicon)

Black coating on mirror is necessary. It should have **small mechanical dissipation** and so on.

Fiber to suspend mirror can be as **thin** as possible (Fiber must support weight of mirror).

Typical diameter is **0.5 mm**.



2. Thermal noise

Below 20K operation (Sapphire, Silicon)

Heat absorbed in mirror : about **1 W**

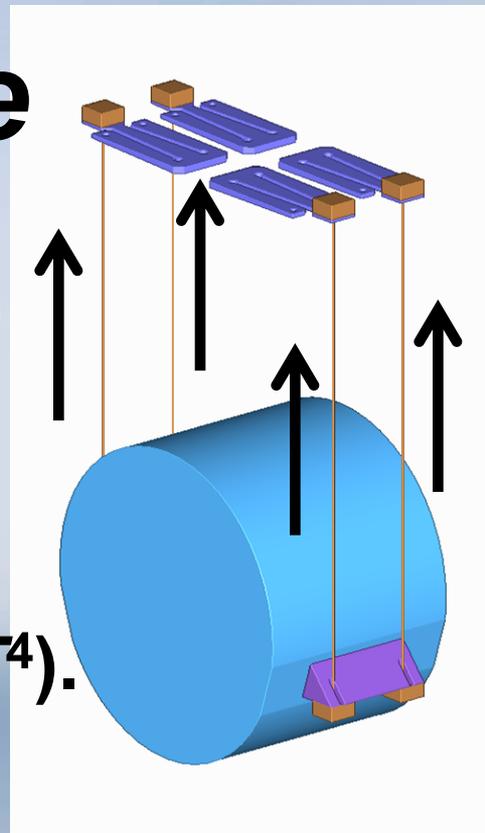
Heat extraction : Radiation is useless (T^4).

Conduction in fibers

Sapphire or silicon has extremely high conductivity (5000 W/m/K).

But, **thicker** fiber for heat conduction (**1.6 mm** in diameter, KAGRA) is necessary.

When heat transfer is not necessary, 0.5 mm in diameter.



2. *Thermal noise*

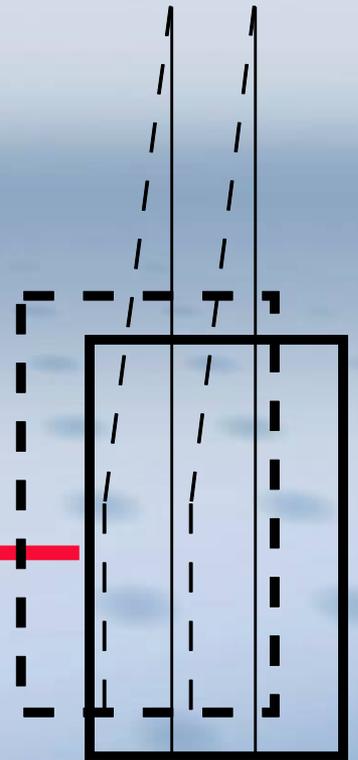
What is an issue of **thicker fiber** ?

Larger pendulum thermal noise

Energy of **pendulum mode**
= gravitational potential energy
+ fiber elastic energy.

Gravitational potential energy has
no loss. **Elastic energy** has **loss**.

Thicker fibers has **larger** elastic energy,
loss, and **thermal noise**.

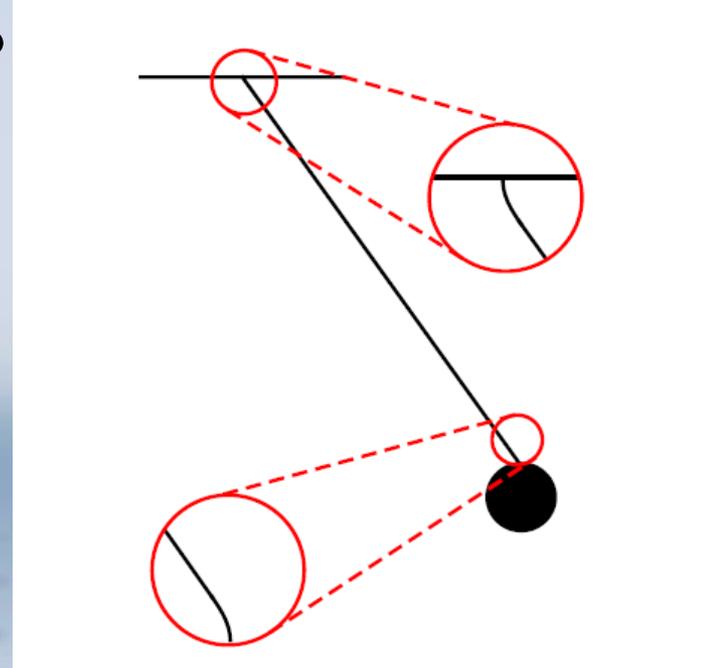


2. Thermal noise

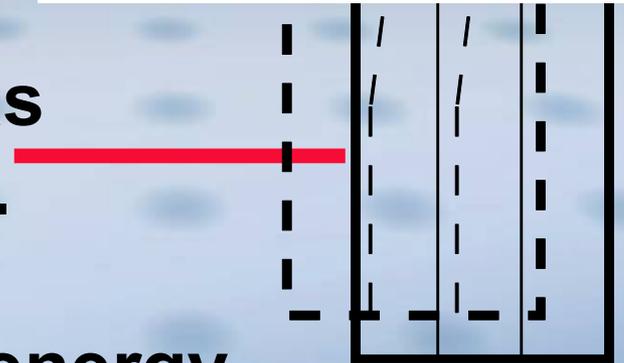
What is an issue of **thicker fiber** ?

Larger suspension thermal noise

Energy of **pendulum mode**
 = gravitational potential energy
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Gravitational potential energy has
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Thicker fibers has **larger** elastic energy,
 loss, and **thermal noise**.

2. *Thermal noise*

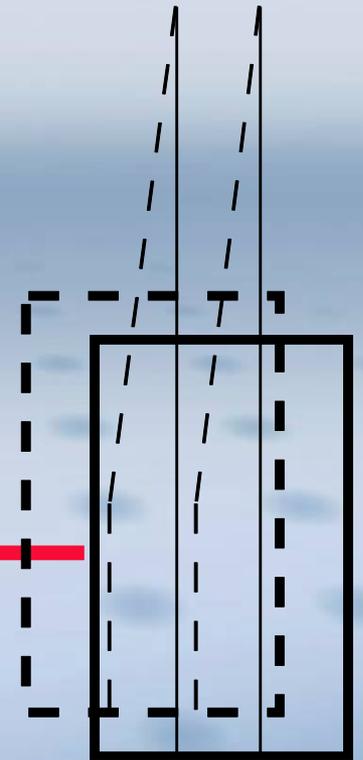
What is an issue of **thicker fiber** ?

Larger pendulum thermal noise

Energy of **pendulum mode**
= gravitational potential energy
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Thicker fibers has **larger** elastic energy,
loss, and **thermal noise**.



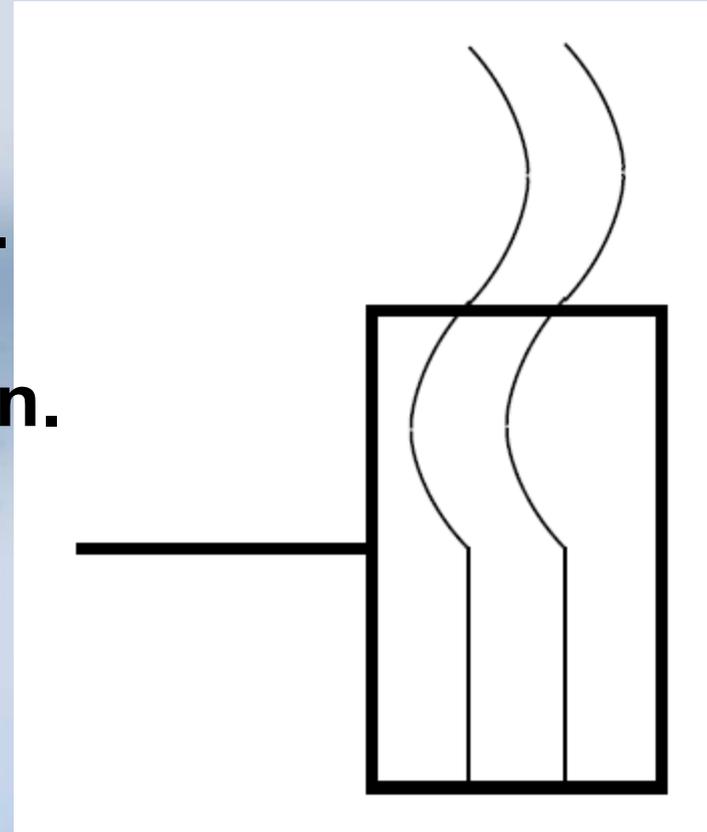
2. *Thermal noise*

What is an issue of **thicker fiber** ?

Thermal noise of **violin mode**

Violin mode : Back action of fiber.

Thicker fiber has large back action.
Mode frequency is **lower**.



2. *Thermal noise*

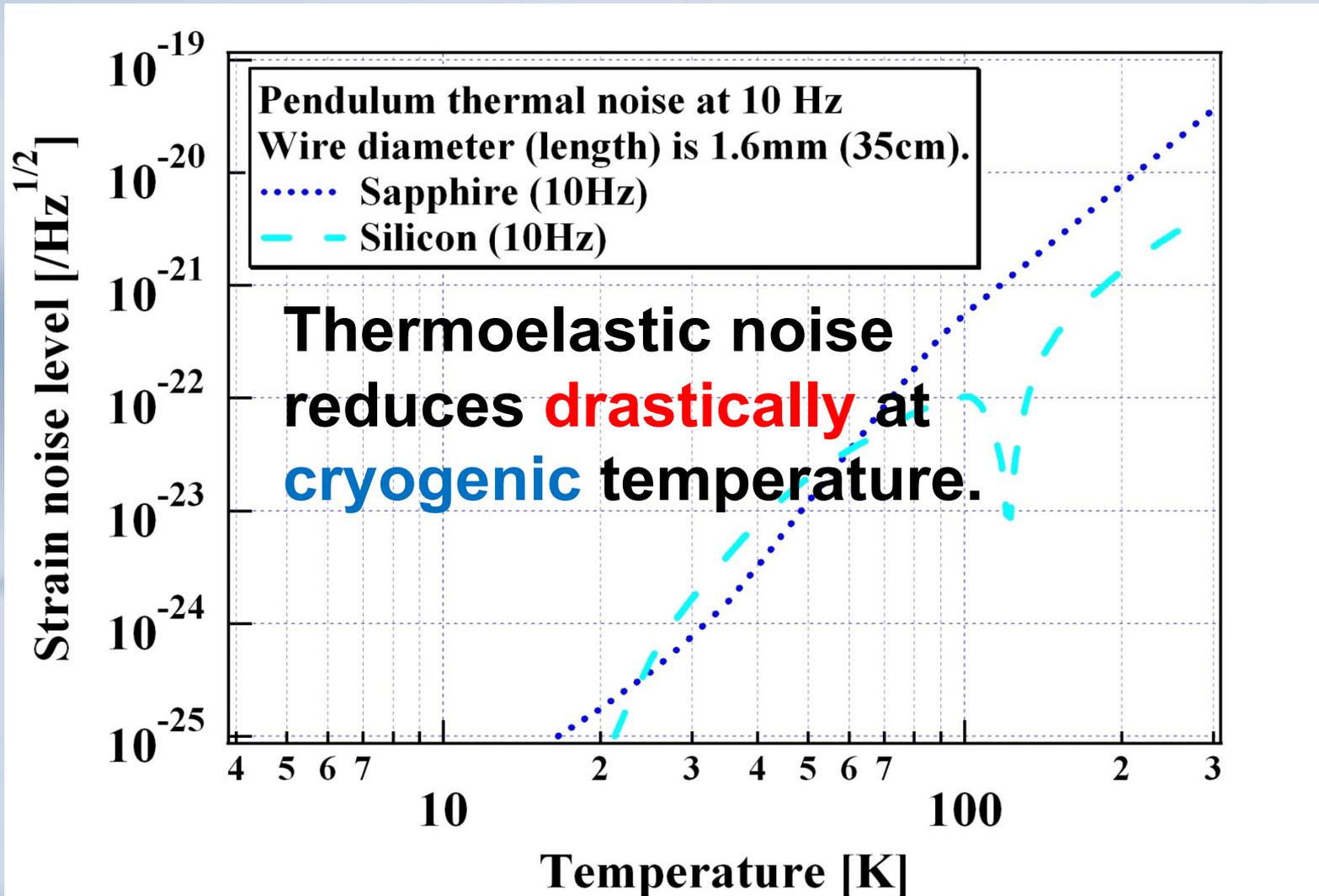
Mechanical dissipation in fibers generates suspension thermal noise.

Two kinds of loss : Structure, Thermoelastic

Note : Both of them in fiber is different from those in 3 dimensional bulk.

2. Thermal noise

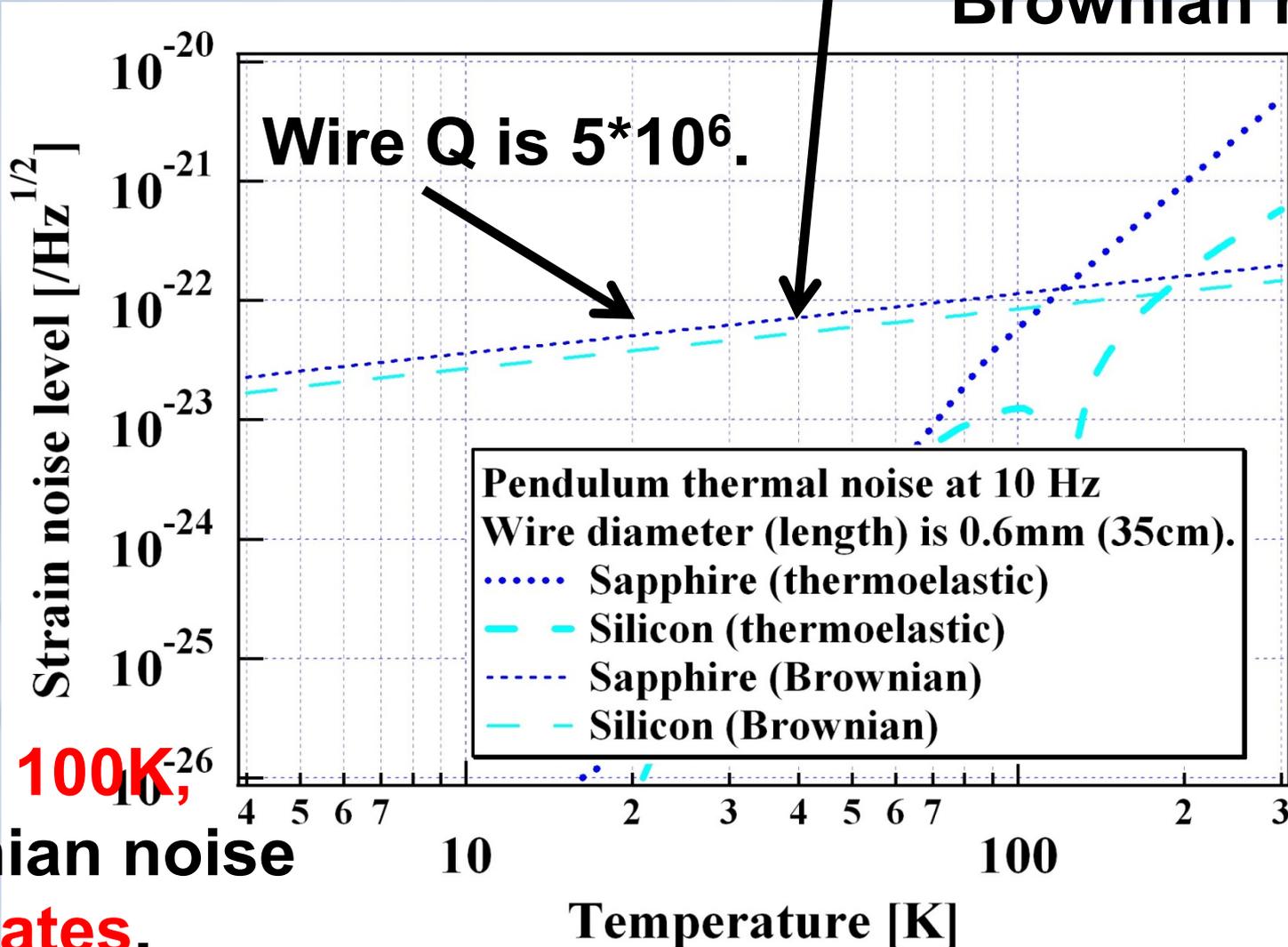
Pendulum thermal noise (Thermoelastic damping)



2. Thermal noise

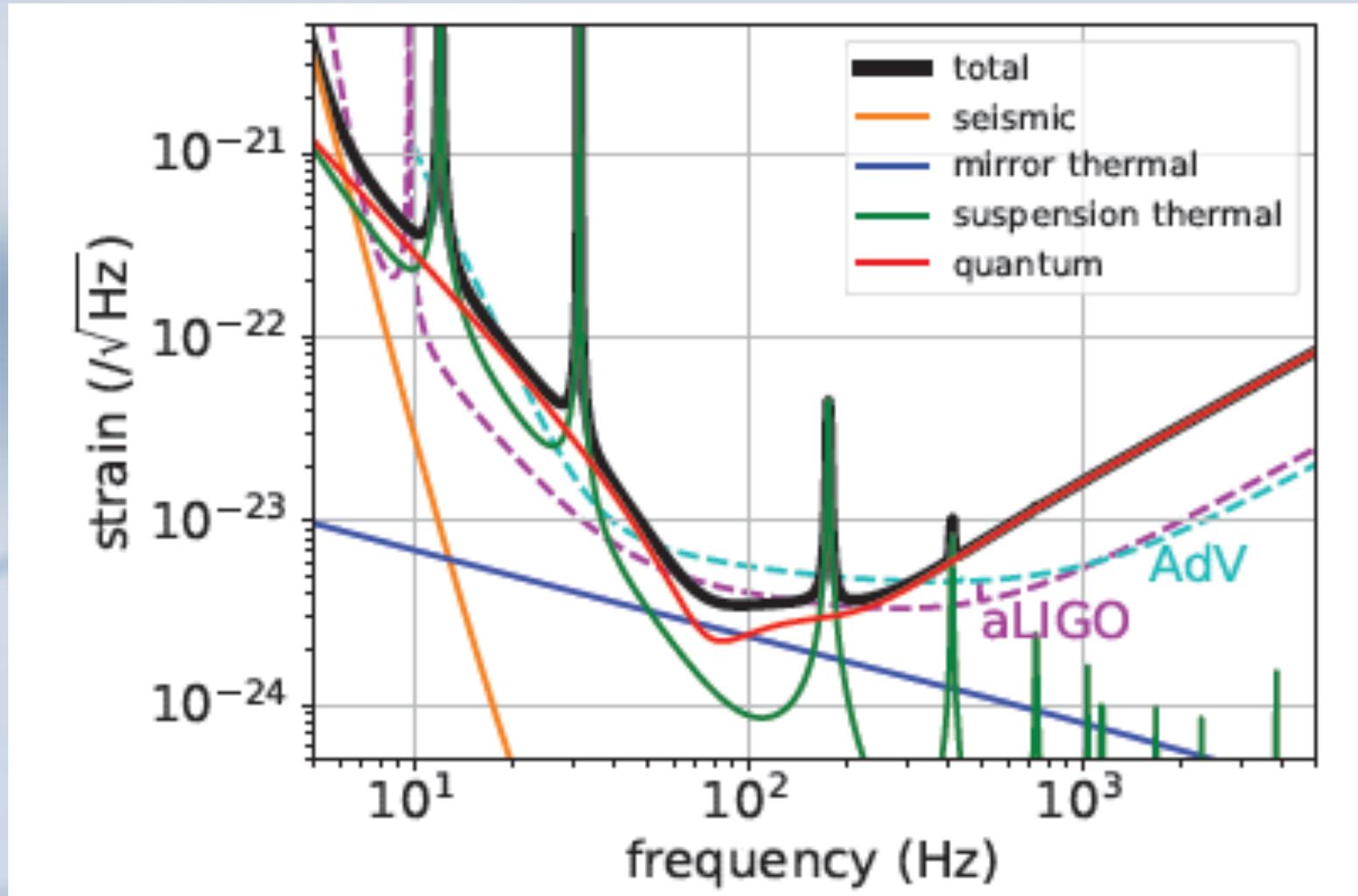
Pendulum thermal noise (Structure damping)

Brownian noise



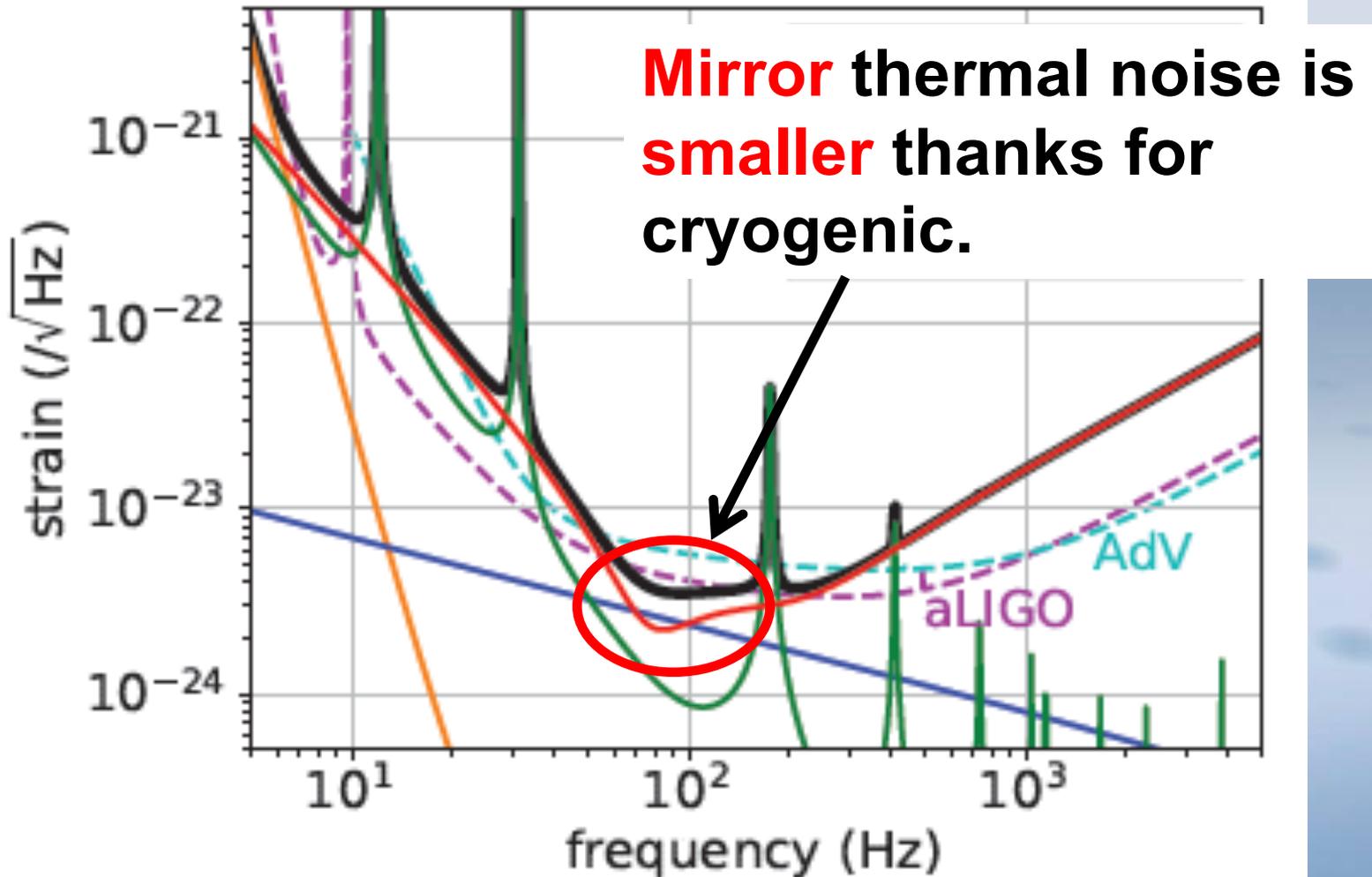
2. Thermal noise

Comparison of KAGRA (Sapphire, 20K) with room temperature interferometer



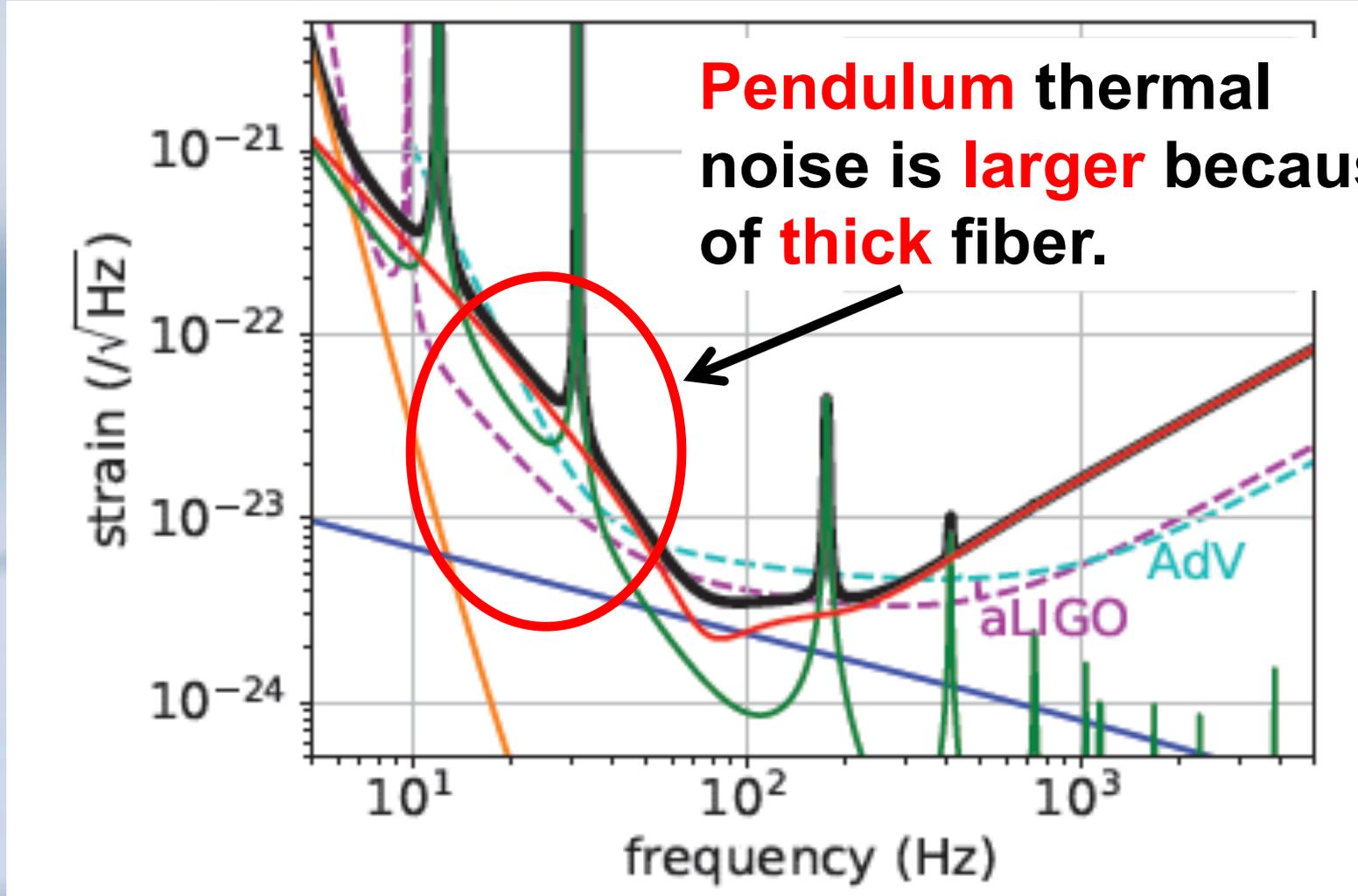
2. Thermal noise

Comparison of KAGRA (Sapphire, 20K) with room temperature interferometer



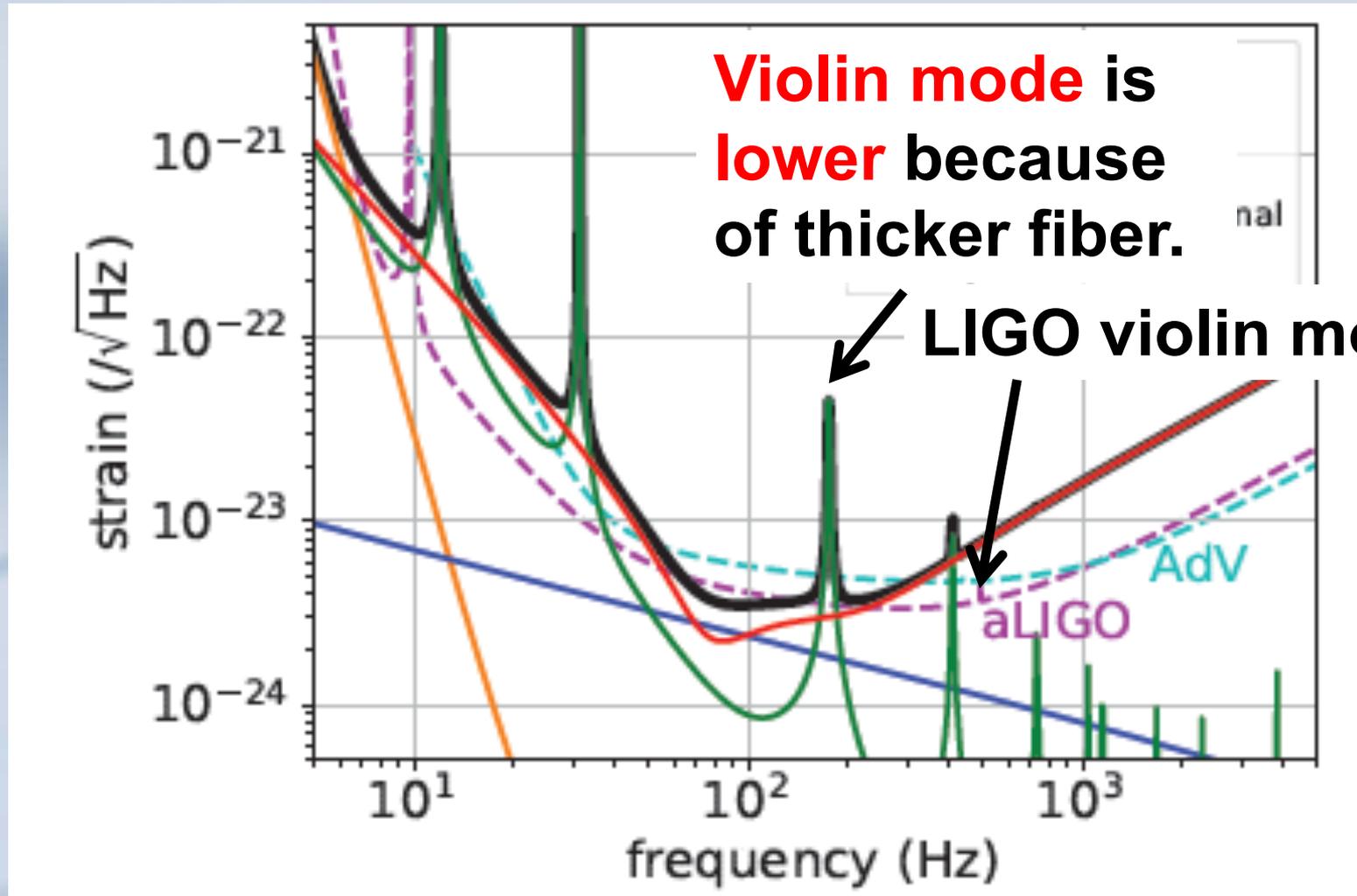
2. Thermal noise

Comparison of KAGRA (Sapphire, 20K) with room temperature interferometer



2. Thermal noise

Comparison of KAGRA (Sapphire, 20K) with room temperature interferometer



2. *Thermal noise*

Sapphire, Silicon, 20K

Mirror thermal noise is **small**.

Suspension (pendulum and violin mode)

thermal noise is an **issue**

(because of thick fiber or heat transmission).

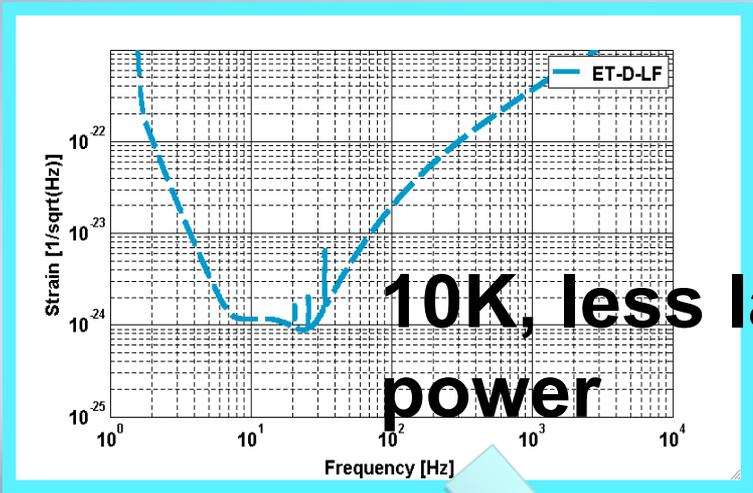
Solution

(1) **Lower frequency** interferometer
with **lower laser power** (Einstein Telescope)

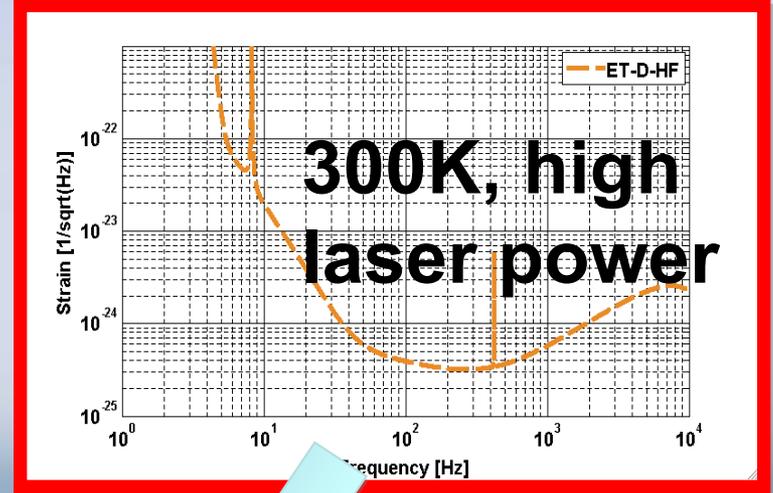
(2) **Small heat absorption** mirror

Large thermal conductivity fibers

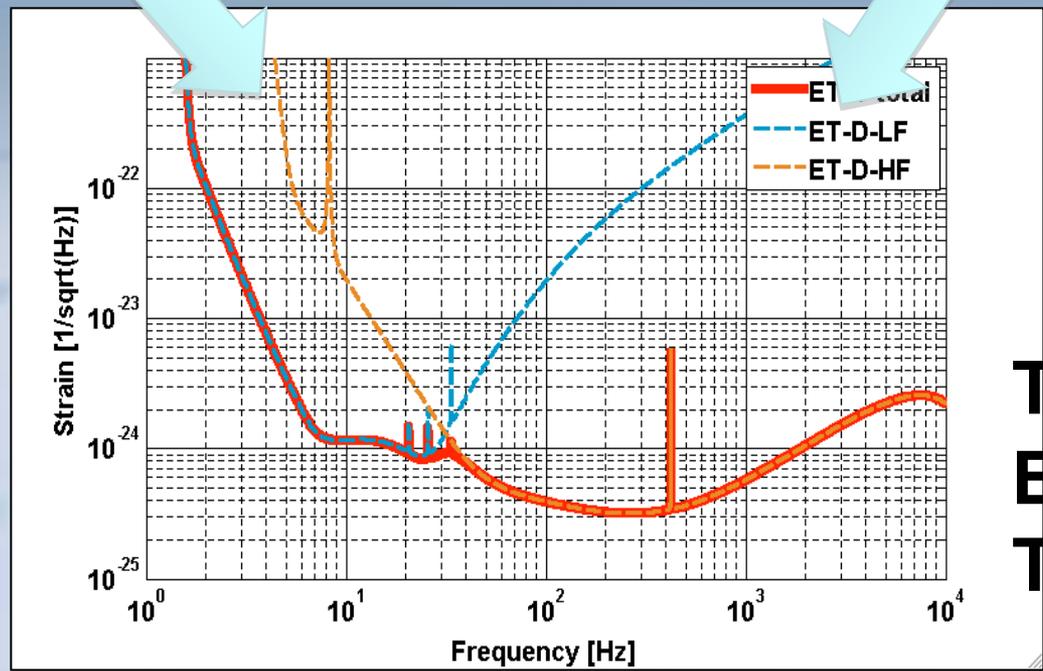
2. Thermal noise



ET-D-LF



ET-D-HF

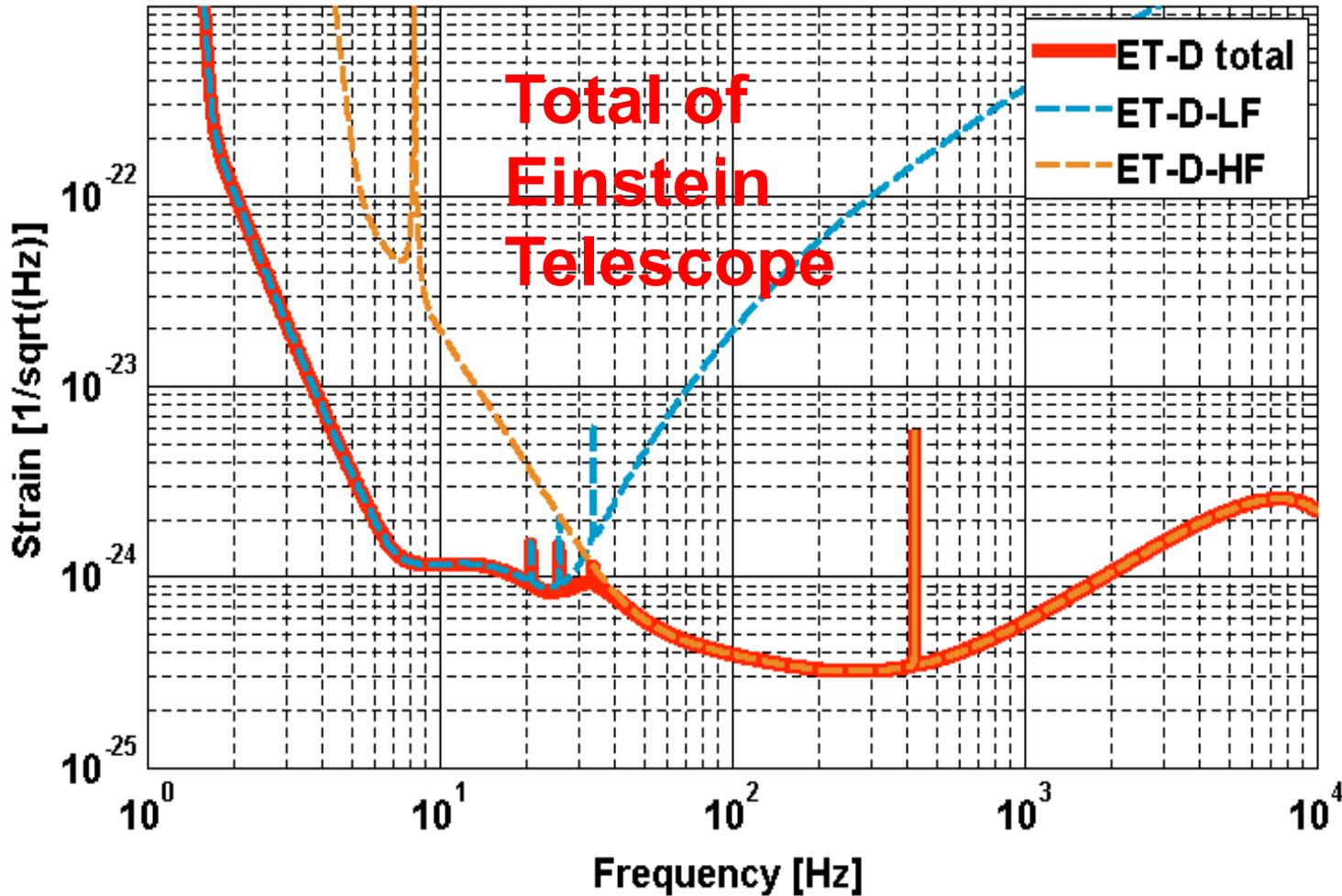


Total of Einstein Telescope

2. Thermal noise

Sapphire, Silicon, 20K

M
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S



on).

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2. Thermal noise

Sapphire, Silicon, 20K

Mirror thermal noise is **small**.

Suspension (pendulum and violin mode)

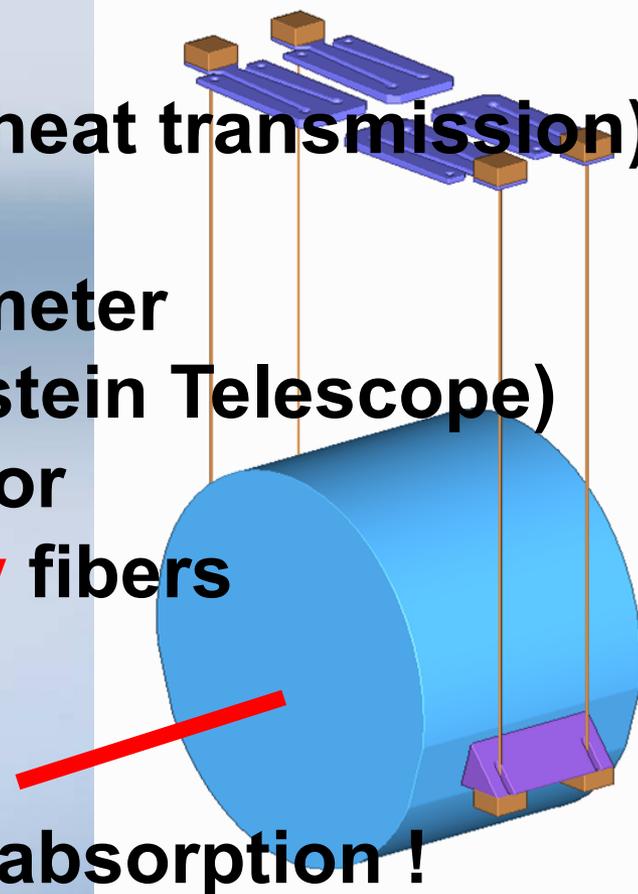
thermal noise is an **issue**

(because of thick fiber or heat transmission).

Solution

(a) **Lower frequency** interferometer
with **lower laser power** (Einstein Telescope)

(b) **Small heat absorption** mirror
Large thermal conductivity fibers



2. *Thermal noise*

(a) **Low heat absorption** in mirror

If heat absorption in mirror (not only substrate but also coating) is **10 - 30 times smaller**, the fiber can be as thin as possible (More thinner fiber can not support mirror).

Substrate : Small absorption (a few ppm/cm) in small sample was **reported**. So not impossible ...

Coating : It is exactly challenge ...

2. *Thermal noise*

(b) Higher thermal conductivity

Size effect : Thermal conductivity is proportional to fiber diameter .

Phonon mean free path is comparable with diameter.

It sounds like **upper limit of thermal conductivity**.

But, ..

2. *Thermal noise*

(b) Higher thermal conductivity

It is assumed that phonon is duffed on fiber surface as like molecules in vacuum duct.

But phonon reflection is **specular** as like light, **thermal conductivity** could be **larger**.



Thermal conductivity (limited by size effect) of silicon is larger when fiber surface is polished .

Phys. Rev. 186, 801 (1969)

3. *Other merits*

Other merits of **cooled** mirror ?

- (1) Thermal lens
- (2) Parametric instability

3. *Other merits*

Thermal lens : Light absorption in mirror

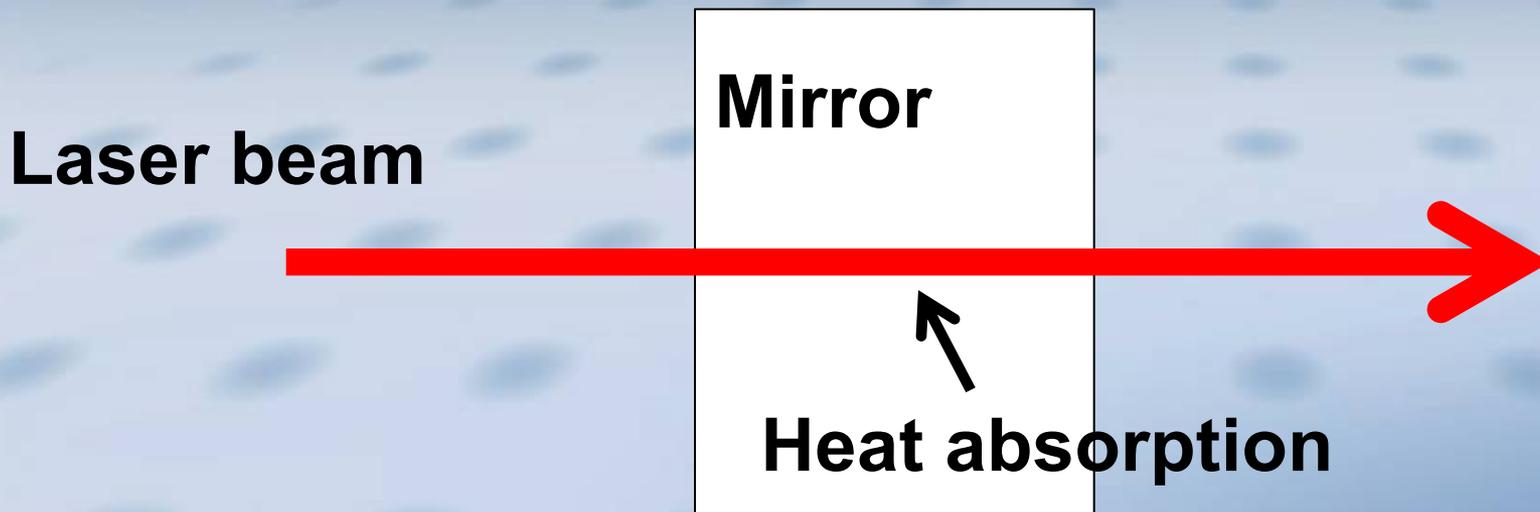
Temperature gradient

Temperature dependent

of refractive index

Wave front distortion

Worse sensitivity



3. *Other merits*

Thermal lens : Light absorption in mirror

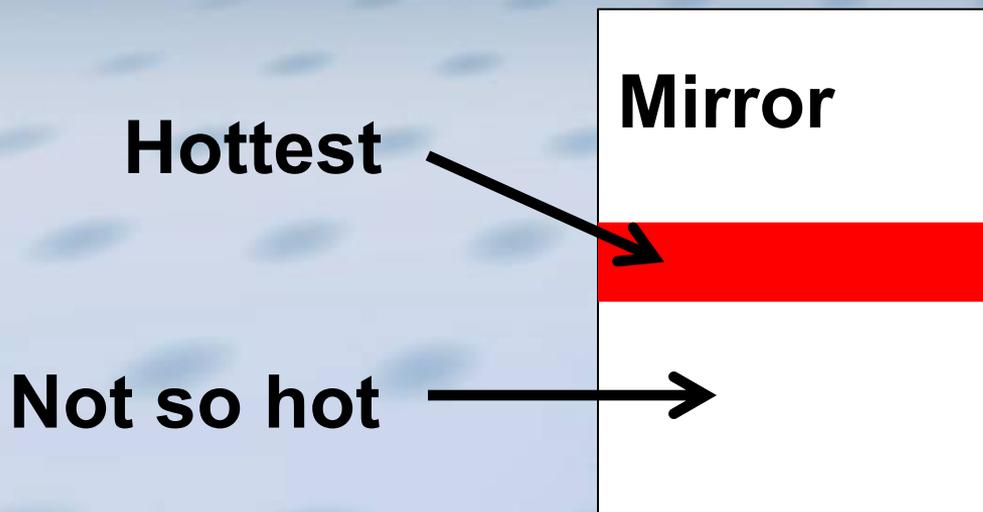
Temperature gradient

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Wave front distortion

Worse sensitivity



3. *Other merits*

Thermal lens : Light absorption in mirror

Temperature gradient

Temperature dependent

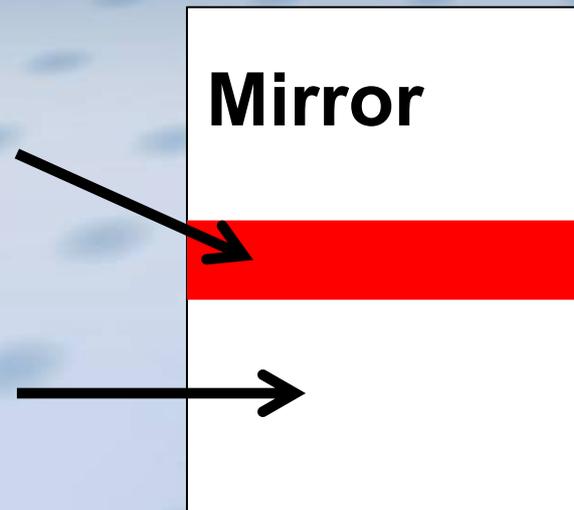
of refractive index

Wave front distortion

Worse sensitivity

Higher or lower
refractive index

Refractive index
does not
change.



Some kind of
imperfect lens !

3. *Other merits*

Thermal lens : Light absorption in mirror

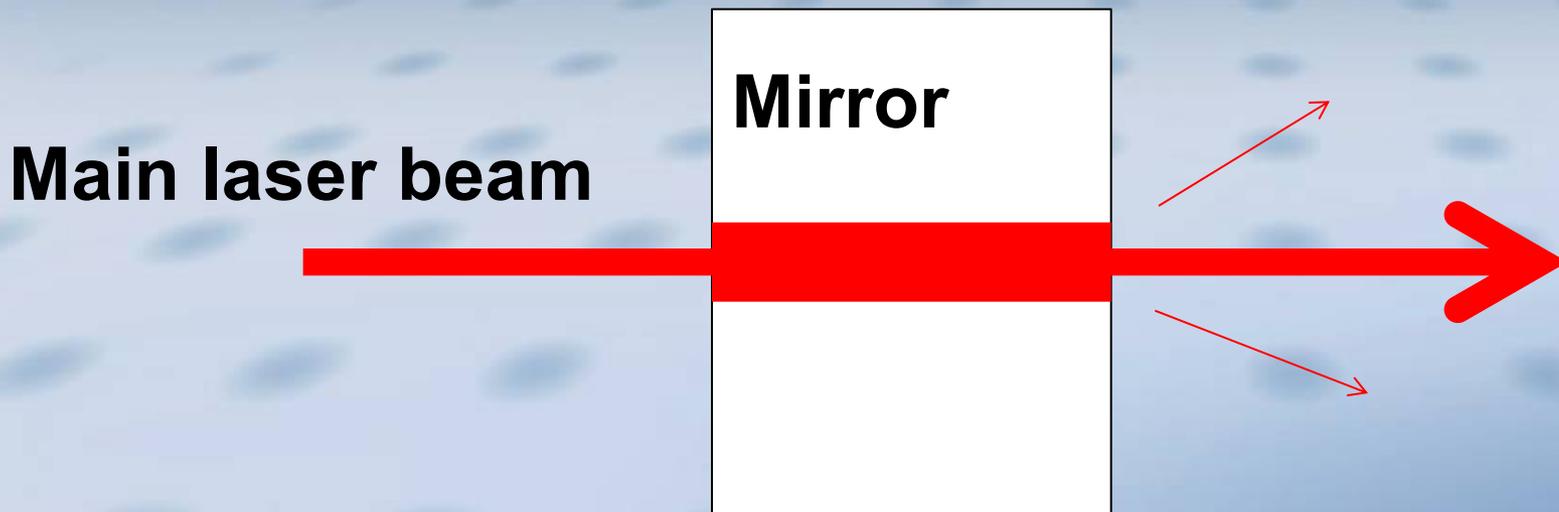
Temperature gradient

Temperature dependent

of refractive index

Wave front distortion

Worse sensitivity



3. *Other merits*

Thermal lens is a **serious problem**
of **room temperature interferometers.**

Advanced LIGO and Virgo : **System to compensate thermal lens** (compensation plate and ring heater) is **necessary.**

G.M. Harry (for LSC), Classical and Quantum Gravity 27 (2010) 084006.

3. *Other merits*

Thermal lens in Sapphire below 20 K

T. Tomaru *et al.*, *Classical and Quantum Gravity* 19 (2002) 2045.

Magnitude of thermal lens : β / κ

Temperature coefficient of refractive index (β)

is at least 100 times smaller.

Thermal conductivity (κ) of sapphire at 20 K

is 10000 times larger

than that of fused silica at 300 K.

Magnitude of thermal lens is at least 10^6 times smaller.

No system for thermal lens compensation is necessary.

3. *Other merits*

Thermal lens in Silicon

In the case of (below) **20K**, thermal lens is **neglected**.

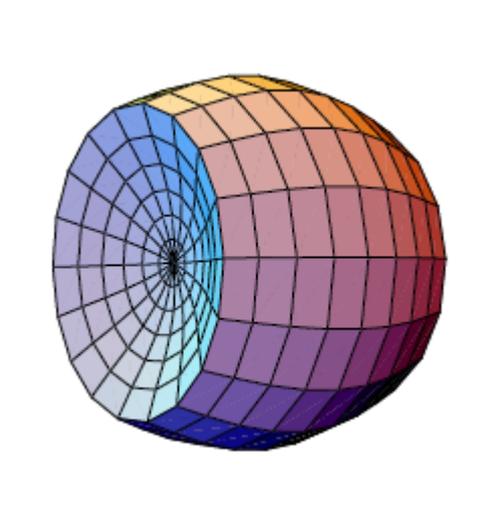
In the case of **120K**, less effective (careful investigation is necessary).

Temperature coefficient of refractive index (β) is **comparable** with that of fused silica at 300 K.

Thermal conductivity (κ) of **silicon at 120 K** is **1000** times larger.

Magnitude of thermal lens is **10^3** times smaller.

3. Other merits



Parametric instability

Radiation pressure

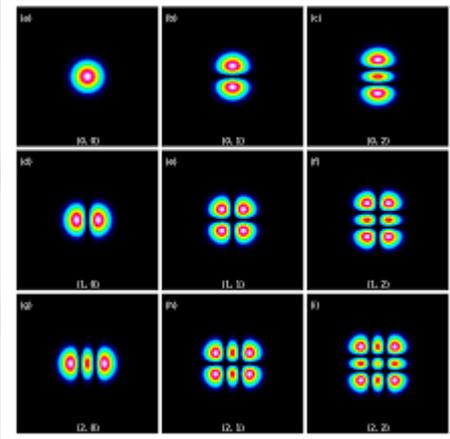


Optical mode in cavity
 (Large amplitude of other (transverse) optical mode)

Elastic mode in mirror
 (Large vibration)



Modulation



3. *Other merits*

Parametric instability of **cryogenic interferometer** is a **less serious problem than that of room temperature.**

K. Yamamoto *et al.*, Journal of Physics: Conference Series 122 (2008) 012015.

(a) **Number of unstable modes is smaller.**

(b) **More effective passive suppression of instability is possible.**

3. *Other merits*

(a) **Number of unstable modes is smaller.**

Number of unstable modes is proportional to the **product of elastic and optical mode densities.**

Elastic mode density is inversely proportional to **cubic of sound velocity.**

Sound velocity in **sapphire** and **silicon** is about **twice times larger** than that in fused silica.

Elastic mode density of sapphire and silicon is **4 or 5 times smaller.**

3. *Other merits*

(a) **Number of unstable modes is smaller.**

Number of unstable modes is proportional to the **product of elastic and optical mode densities.**

Optical mode density is **smaller** when **beam radius is smaller.**

Larger beam radius is one of techniques to **reduce mirror thermal noise.**

Cryogenic interferometer can adopt **smaller beam** because of small mirror thermal noise.

3. *Other merits*

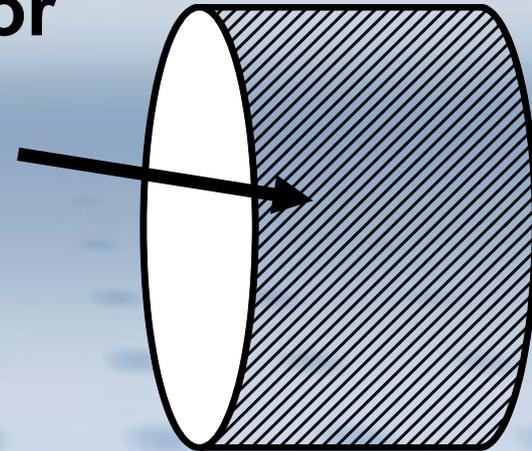
(b) **More effective passive suppression**

of instability is possible.

Passive suppression :

loss on barrel surface of mirror

loss on barrel surface



The **increase of thermal noise** should be taken into account.

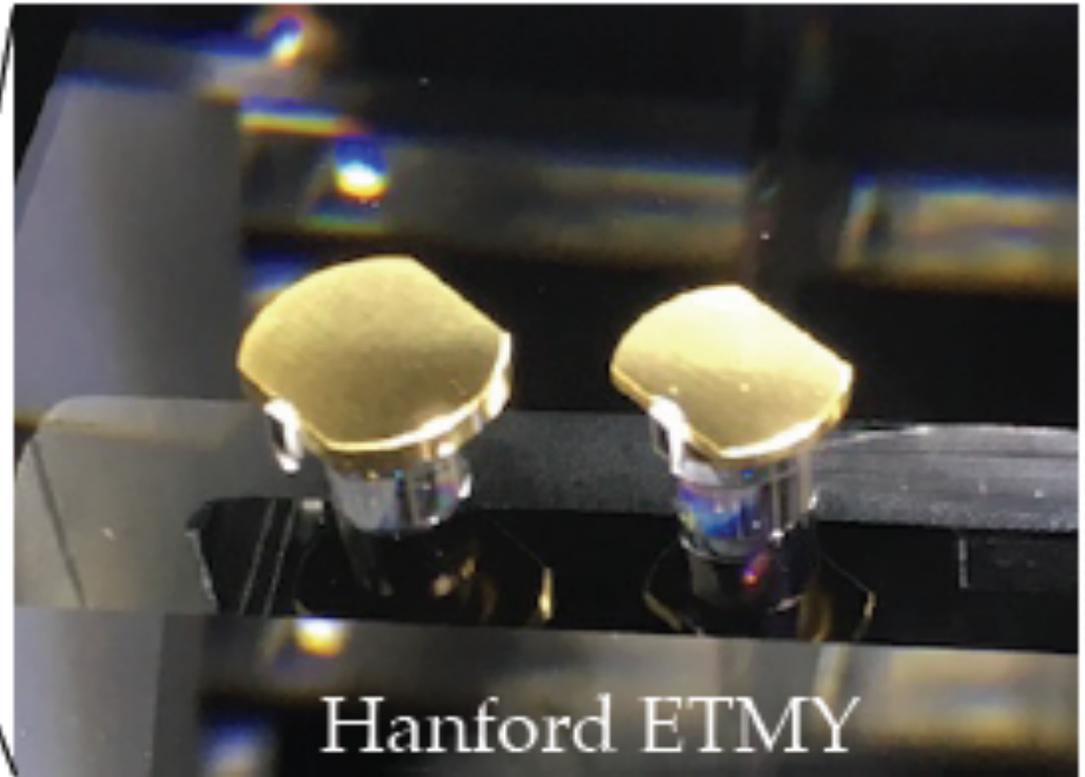
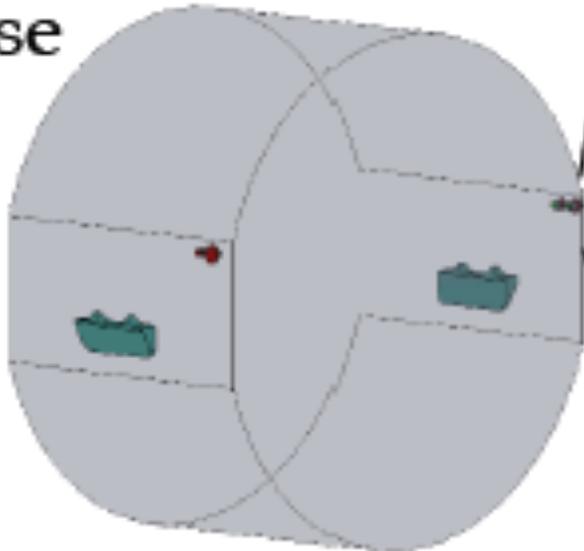
In the case of **cooled** mirror, more effective suppression is **possible**.

3. Other merits

(b) More effective passive suppression

P LIGO

ase



LHO alog 41231

In the case of **coated** mirror, more effective suppression is **possible**.

3. *Other merits*

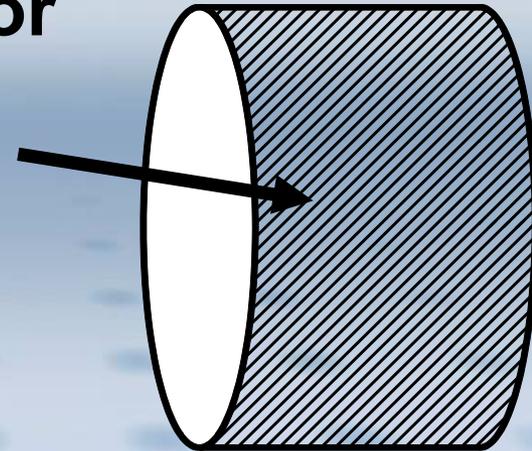
(b) **More effective passive suppression**

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Passive suppression :

loss on barrel surface of mirror

loss on barrel surface



The **increase of thermal noise** should be taken into account.

In the case of **cooled** mirror, more effective suppression is **possible**.

4. How to cool down

How to cool down **actually** ? (KAGRA bias ...)

(1) **Cooling bath** (or something similar)

is necessary.

Cooling power is as large as possible.

(2) **Temperature** after cooling down is

as **low** as possible.

Heat load (except for mirror absorption)

must be negligible.

(120K operation; temperature adjustment system is necessary. I skip this topic).

(3) **Initial cooling time** is as **short** as possible for long observation time.

4. *How to cool down*

(1) **Cooling bath** (or something similar)

Liquid helium, nitrogen ...

Large cooling power

No vibration (after cool down)

120K : Only liquid nitrogen is necessary and reuse system is not necessary !

So, this could be a solution

4. *How to cool down*

(1) **Cooling bath** (or something similar)

Discussion in KAGRA (**20K**)

Reuse of helium (cycle of liquid and gas of helium):

Apparatus with high pressure is necessary.

It needs approval by government and license in Japan.

Cost of such apparatus is quite high and complicate.

Liquid helium **in tunnel** is not so fun

4. How to cool down

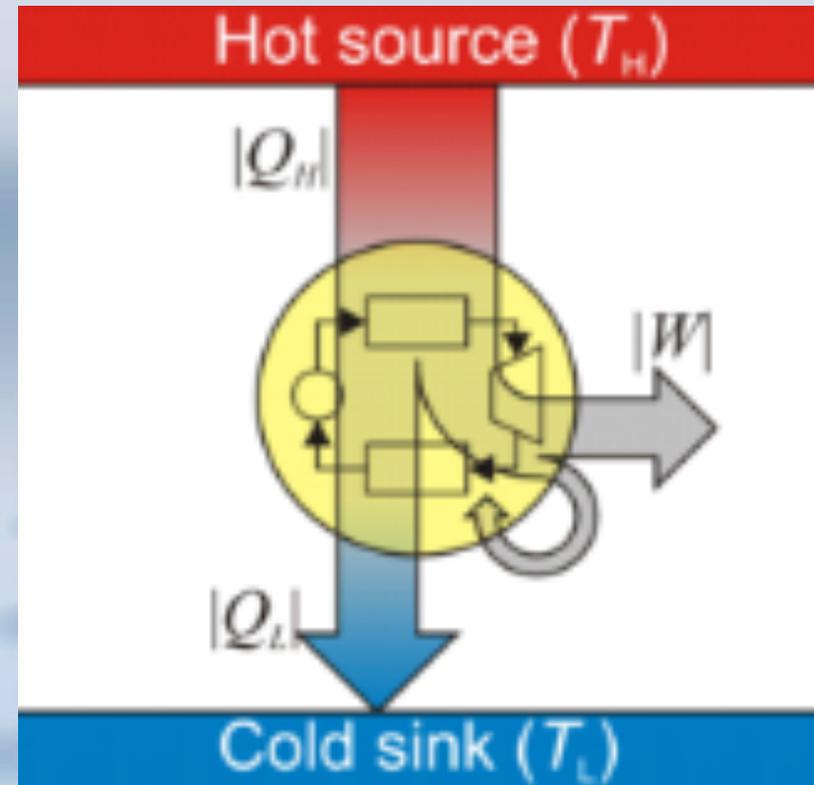
(1) **Cooling bath** (or something similar) for **below 20K** operation

Cryocooler !

Heat engine

Heat (Q) goes from **hot** to **cold** heat bath.

Part of heat ($Q_H - Q_L$) is **converted** into work (W).



Wikipedia

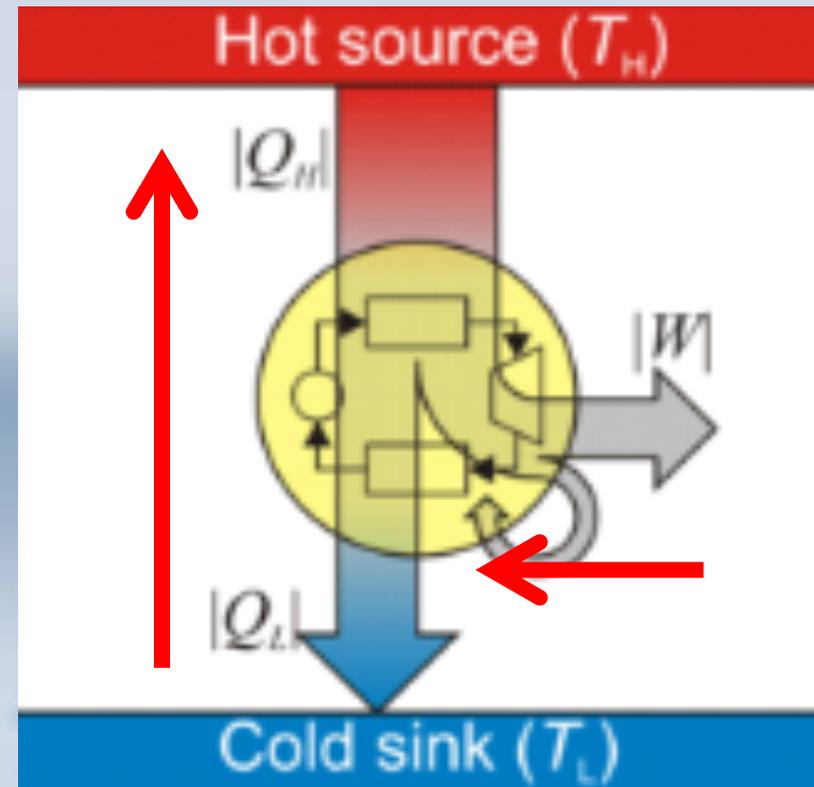
4. How to cool down

(1) **Cooling bath** (or something similar) for **below 20K** operation

Cryocooler !

Work (W) is applied on
cryocooler .

Heat (Q) goes from **cold** to
hot heat bath.



Note : Efficiently (cooling power) at **lower temperature** is **smaller** because of second law of thermodynamics.

4. *How to cool down*

(1) **Cooling bath** (or something similar)

Cryocooler : Reverse operation of heat engine

(Rigid and heavy) Piston motion generates vibration.

Pulse tube cryocooler : “virtual” piston by valve unit.
This “virtual” piston consists of gas !
Small vibration is generated.

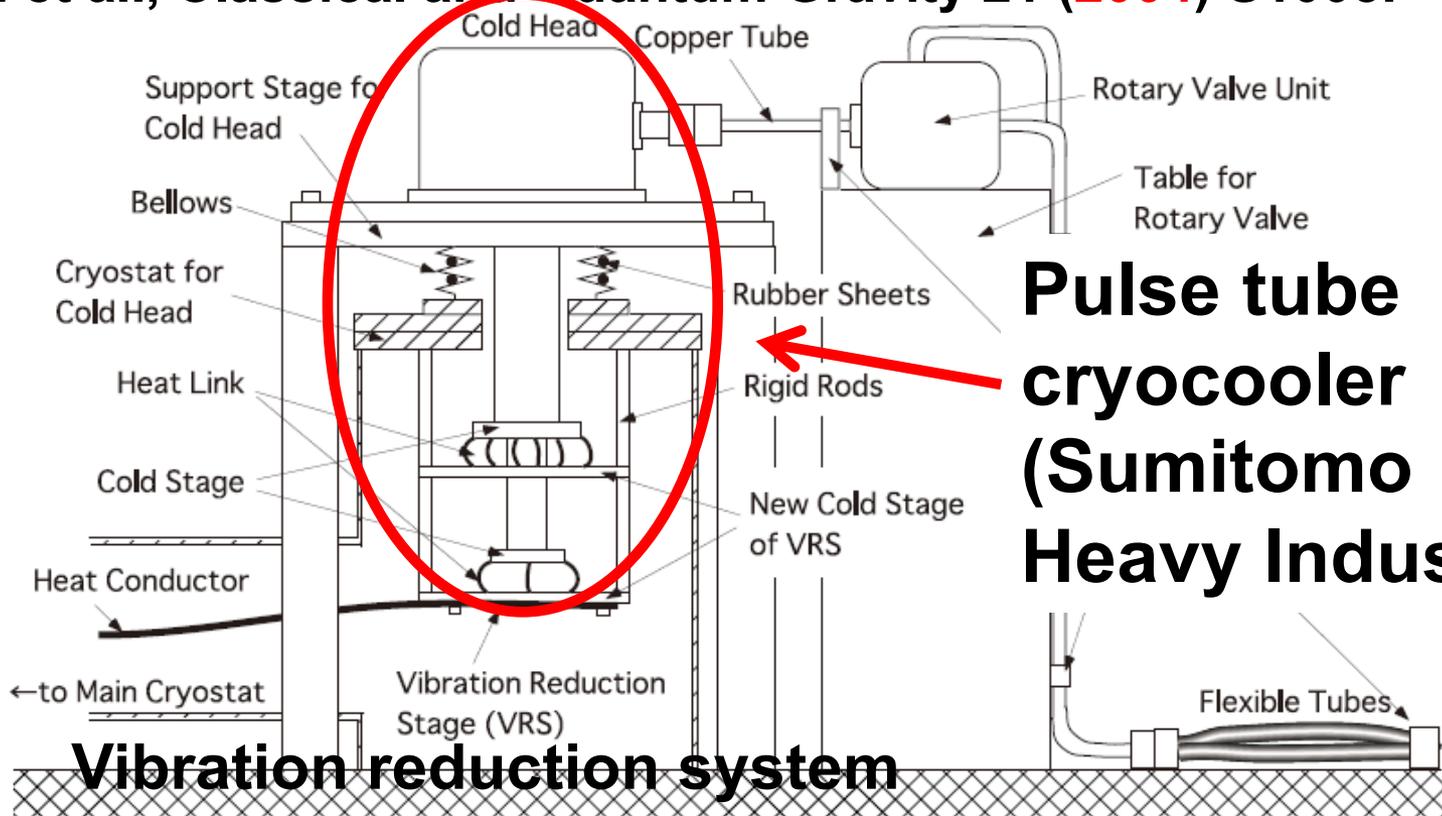
Pulse tube cryocooler : **50 W** (above 60K)
1W (around 8 K)

C. Tokoku *et al.*, CEC/ICMC2013, 2EPoE1-03, Anchorage, USA (2013).

4. How to cool down

KAGRA adopts **pulse tube cryocooler**. Vibration reduction system was developed by us.

T. Tomaru et al., Classical and Quantum Gravity 21 (2004) S1005.



Pulse tube cryocooler (Sumitomo Heavy Industries)

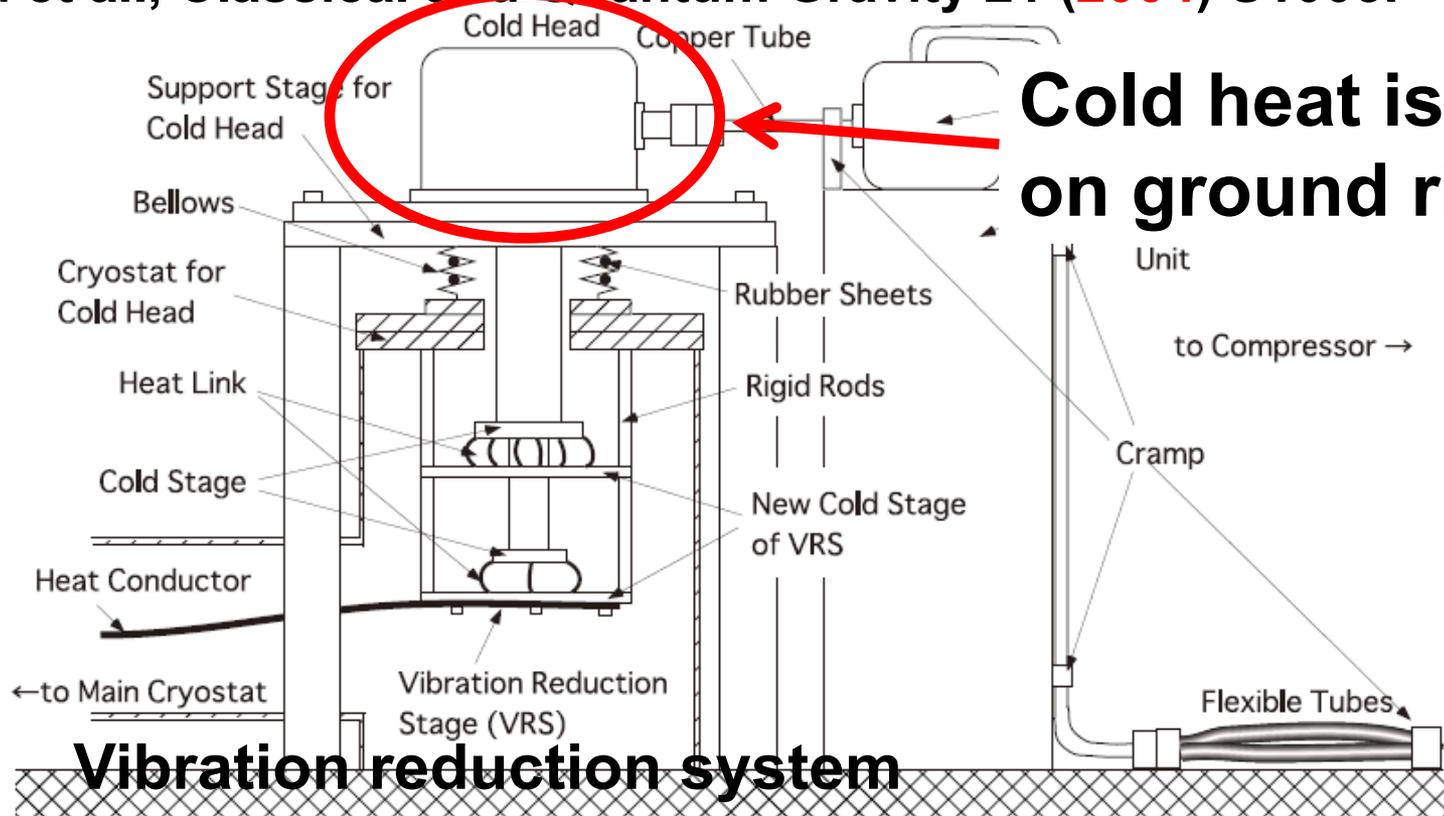
Vibration reduction system

Figure 3. Vibration-reduction system we have been developing for the PT cryocooler.

4. How to cool down

KAGRA adopts **pulse tube cryocooler**. Vibration reduction system was developed by us.

T. Tomaru et al., Classical and Quantum Gravity 21 (2004) S1005.



Cold heat is fixed on ground rigidly.

Vibration reduction system

Figure 3. Vibration-reduction system we have been developing for the PT cryocooler.

4. How to cool down

KAGRA adopts **pulse tube cryocooler**. Vibration reduction system was developed by us.

T. Tomaru et al., Classical and Quantum Gravity 21 (2004) S1005.

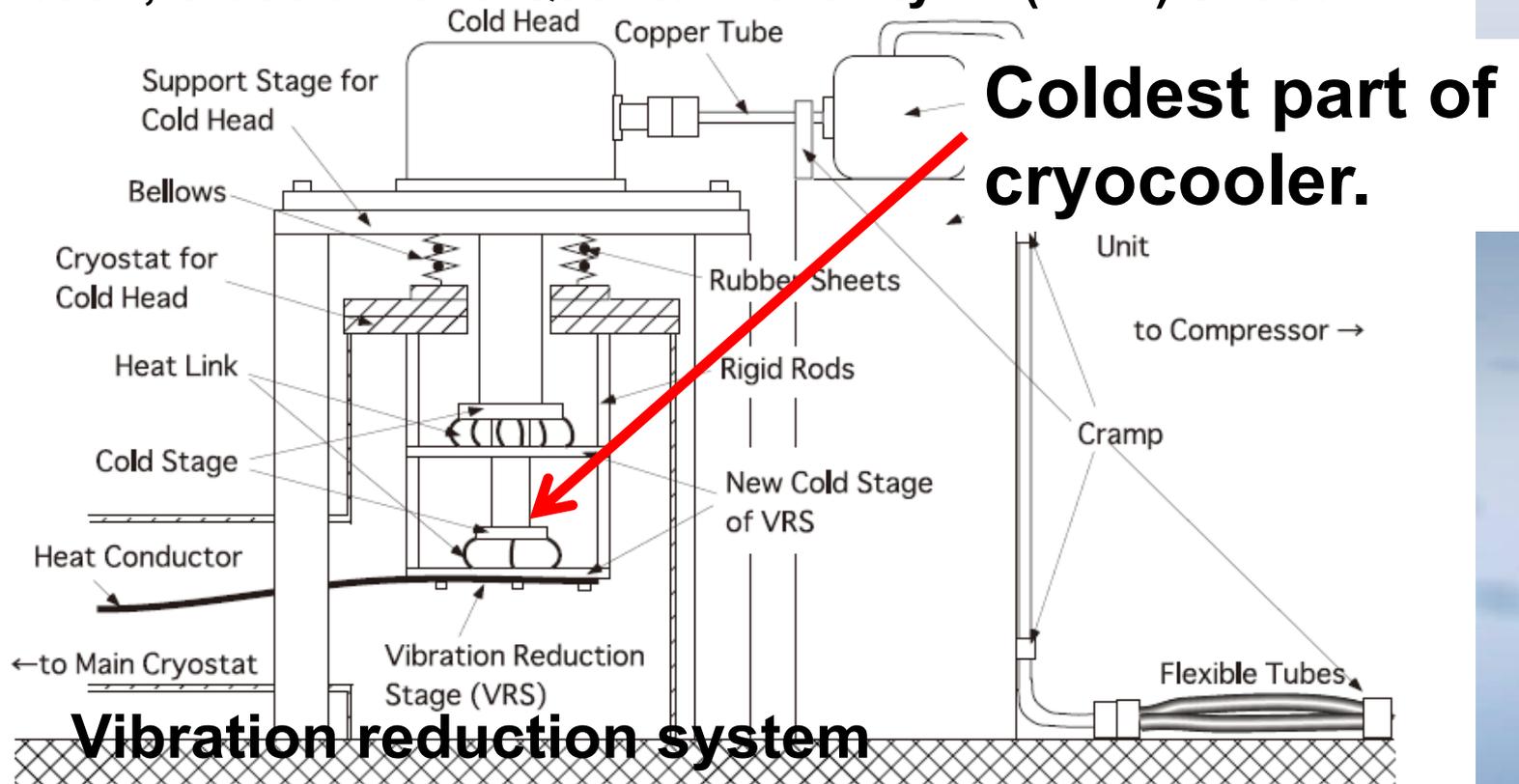


Figure 3. Vibration-reduction system we have been developing for the PT cryocooler.

4. How to cool down

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T. Tomaru et al., Classical and Quantum Gravity 21 (2004) S1005.

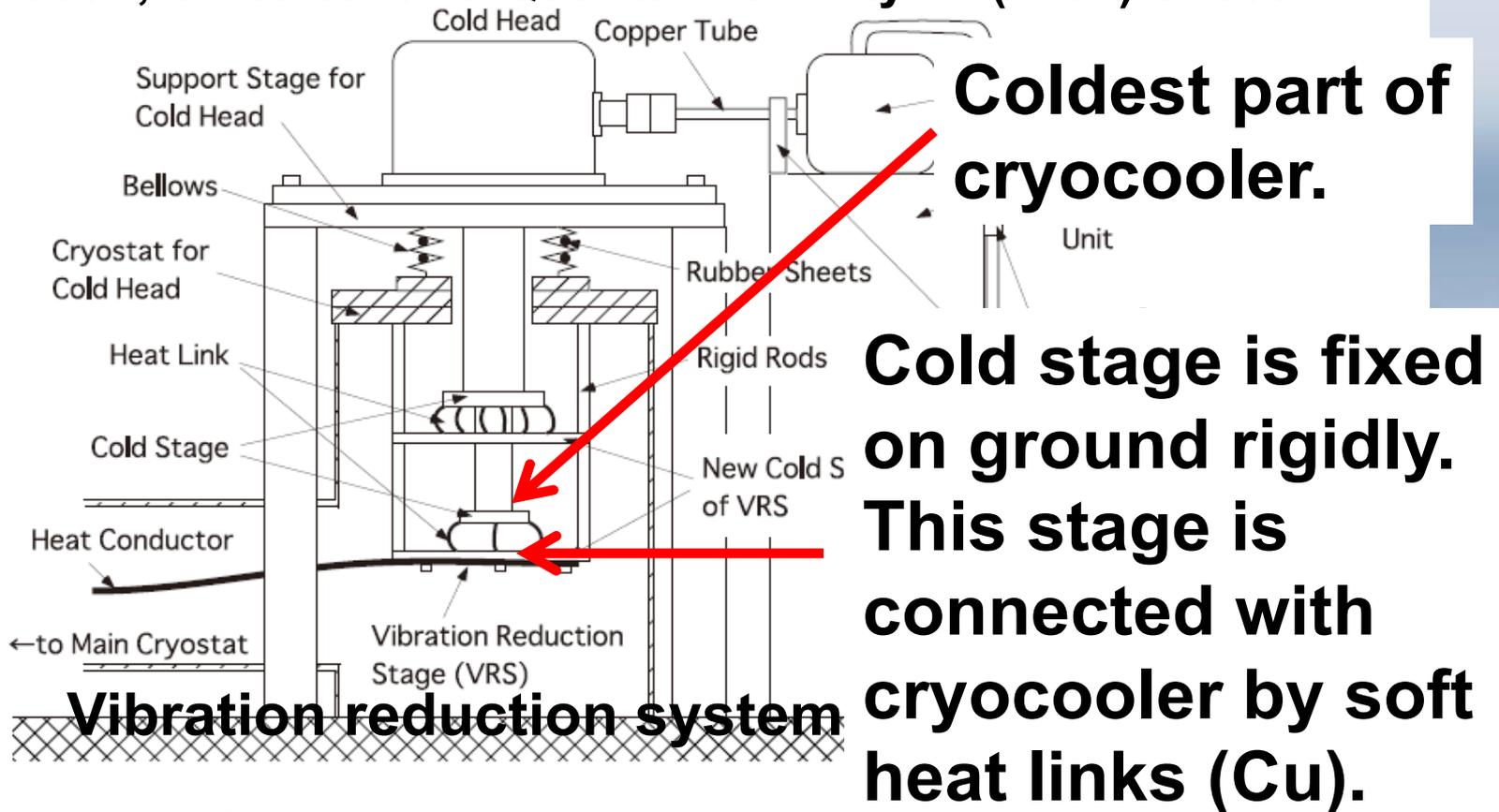
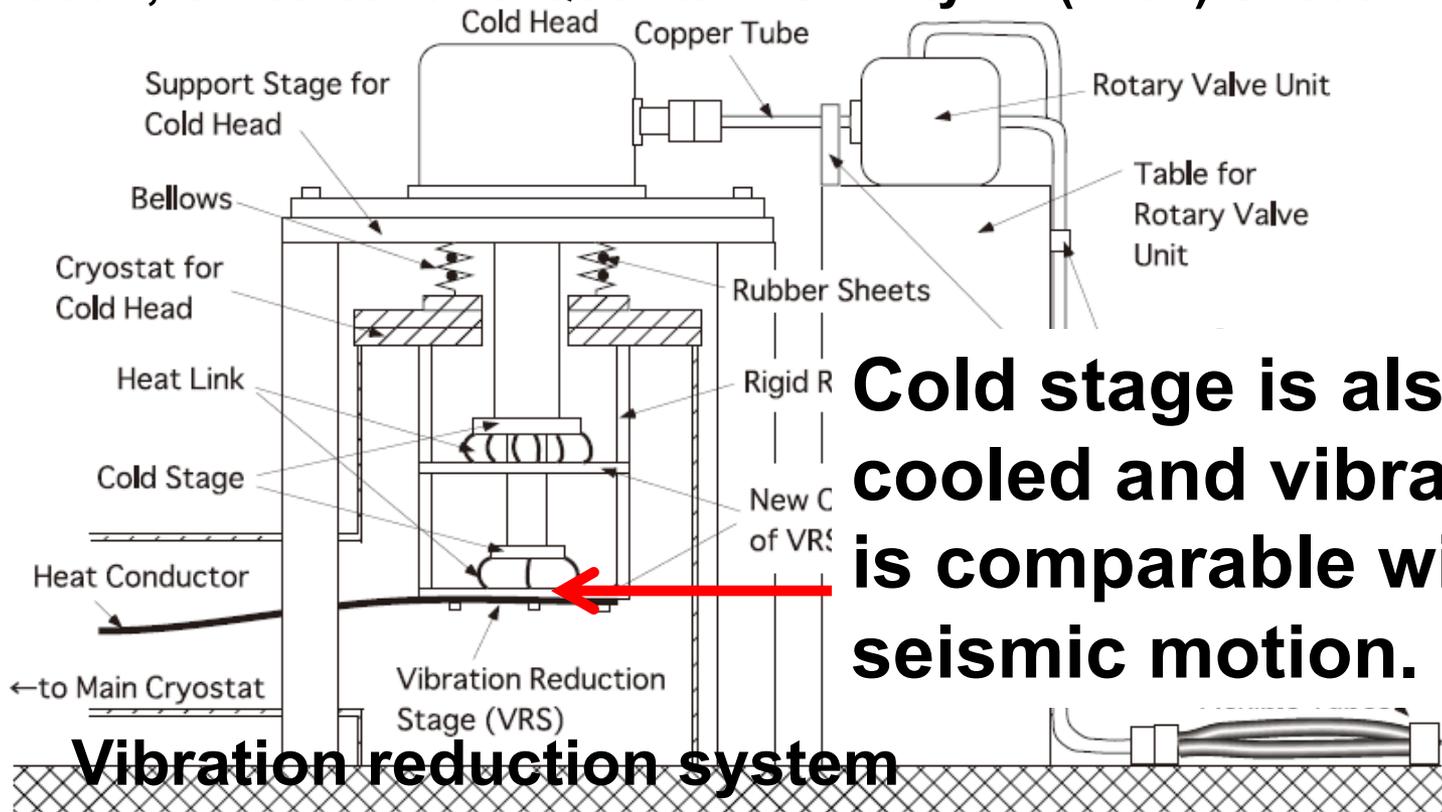


Figure 3. Vibration-reduction system we have been d.....

4. How to cool down

KAGRA adopts **pulse tube cryocooler**. Vibration reduction system was developed by us.

T. Tomaru et al., Classical and Quantum Gravity 21 (2004) S1005.



Cold stage is also cooled and vibration is comparable with seismic motion.

Vibration reduction system

Figure 3. Vibration-reduction system we have been developing for the PT cryocooler.

4. How to cool down

KAGRA adopts **pulse tube cryocooler**. Vibration reduction system was developed by us.

T. Tomaru et al., Classical and Quantum Gravity 21 (2004) S1005.

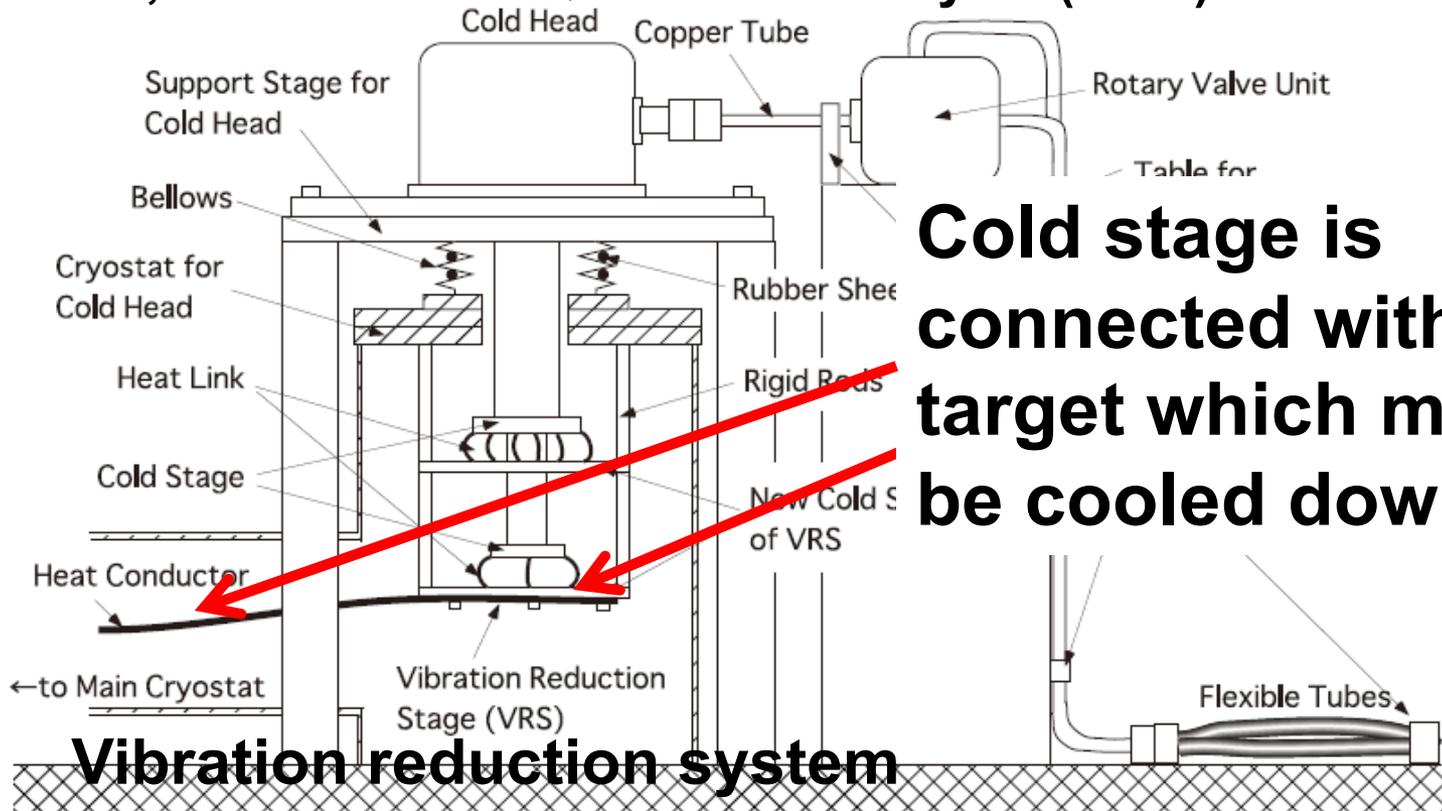


Figure 3. Vibration-reduction system we have been developing for the PT cryocooler.

4. How to cool down

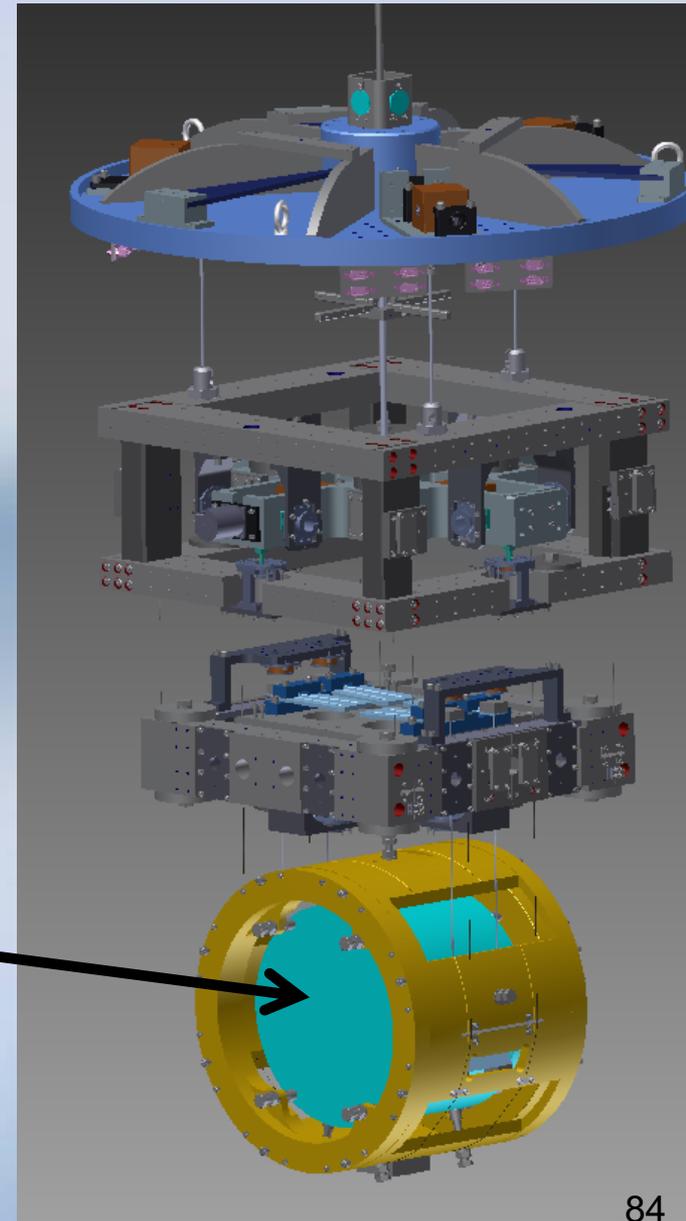
What is cooled down ?

We must cool not only mirrors and fibers **but also** ...

(1) Cryogenic payload : **Control** of mirror position and alignment for light interference (**high sensitivity**).

K. Okutomi,
Wednesday afternoon

Mirror



4. How to cool down

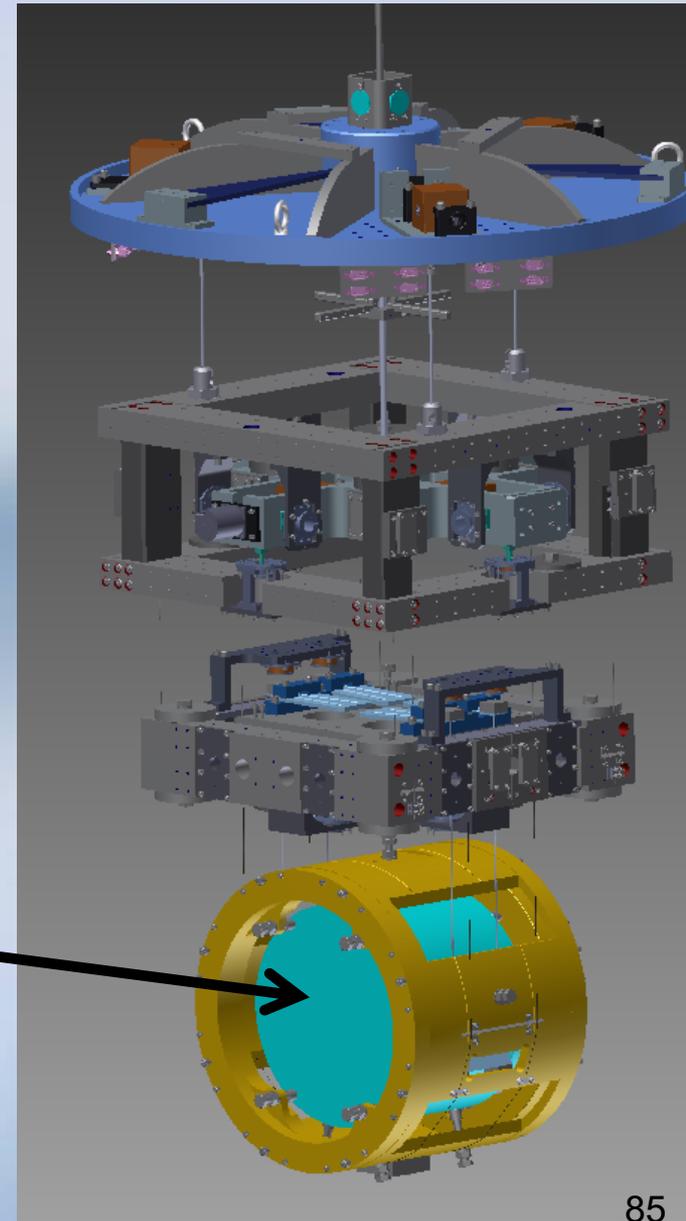
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Wednesday afternoon

Mirror



4. How to cool down

What is cooled down ?

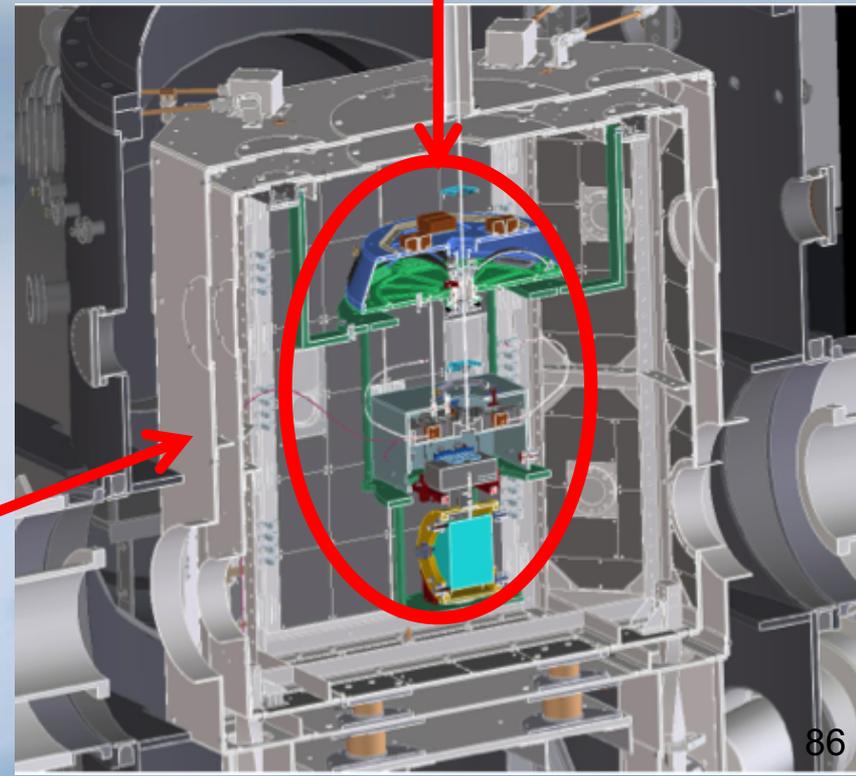
We must cool not only mirrors and fibers **but also**

...

(2) Radiation shield: **Cooled wall** to avoid attack of 300 K on cryogenic payload.

Cryogenic payload

Radiation shield



4. How to cool down

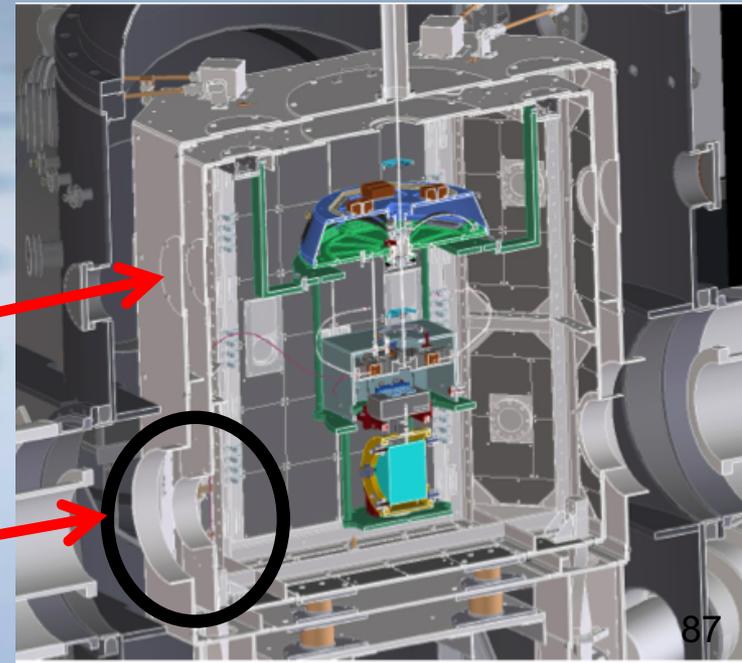
(2) Small heat load

Actually, extra heat load is always issues ...

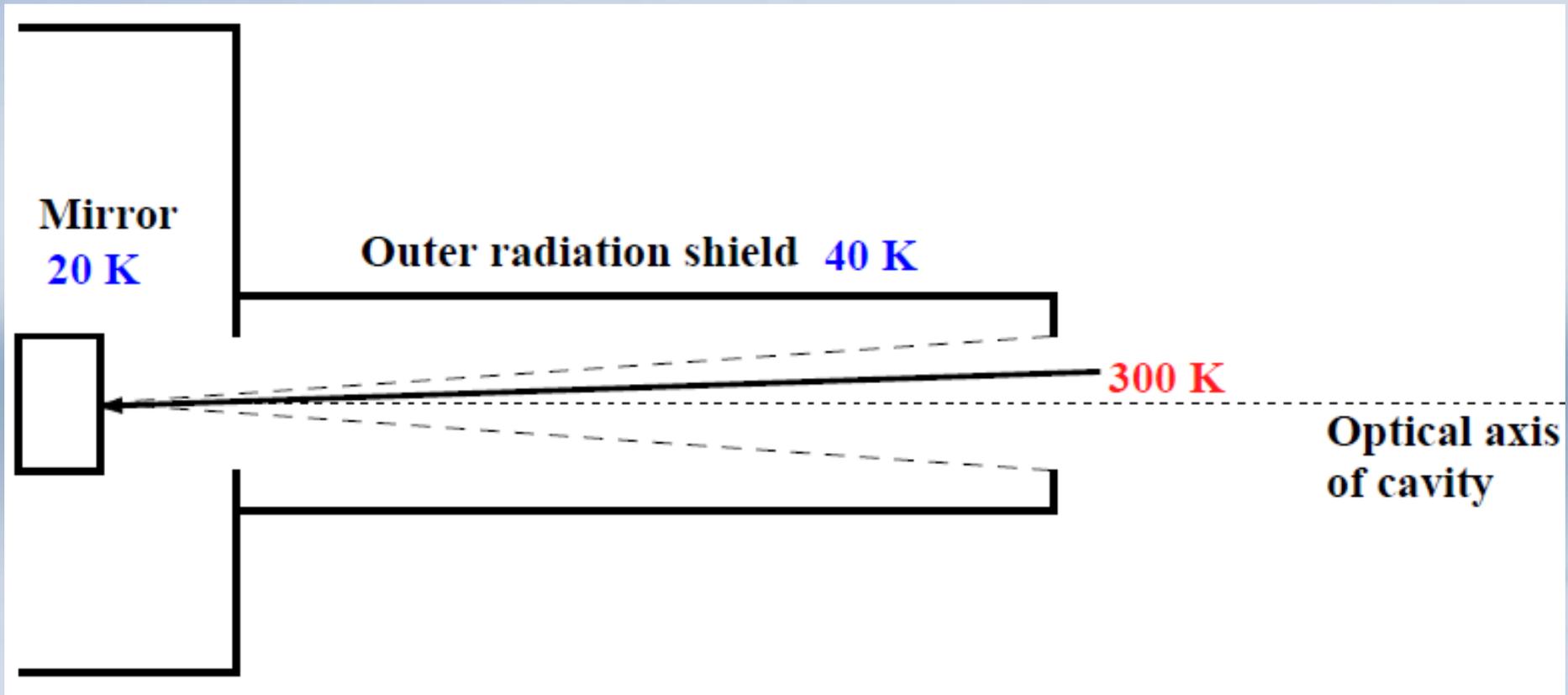
Largest one : **300K radiation through hole for laser beam**. This hole is comparable with mirror itself; 20cm in diameter. 300K radiation is about **20 W** ! It is larger than cryocooler power around 8K!

Radiation shield

Hole for laser beam

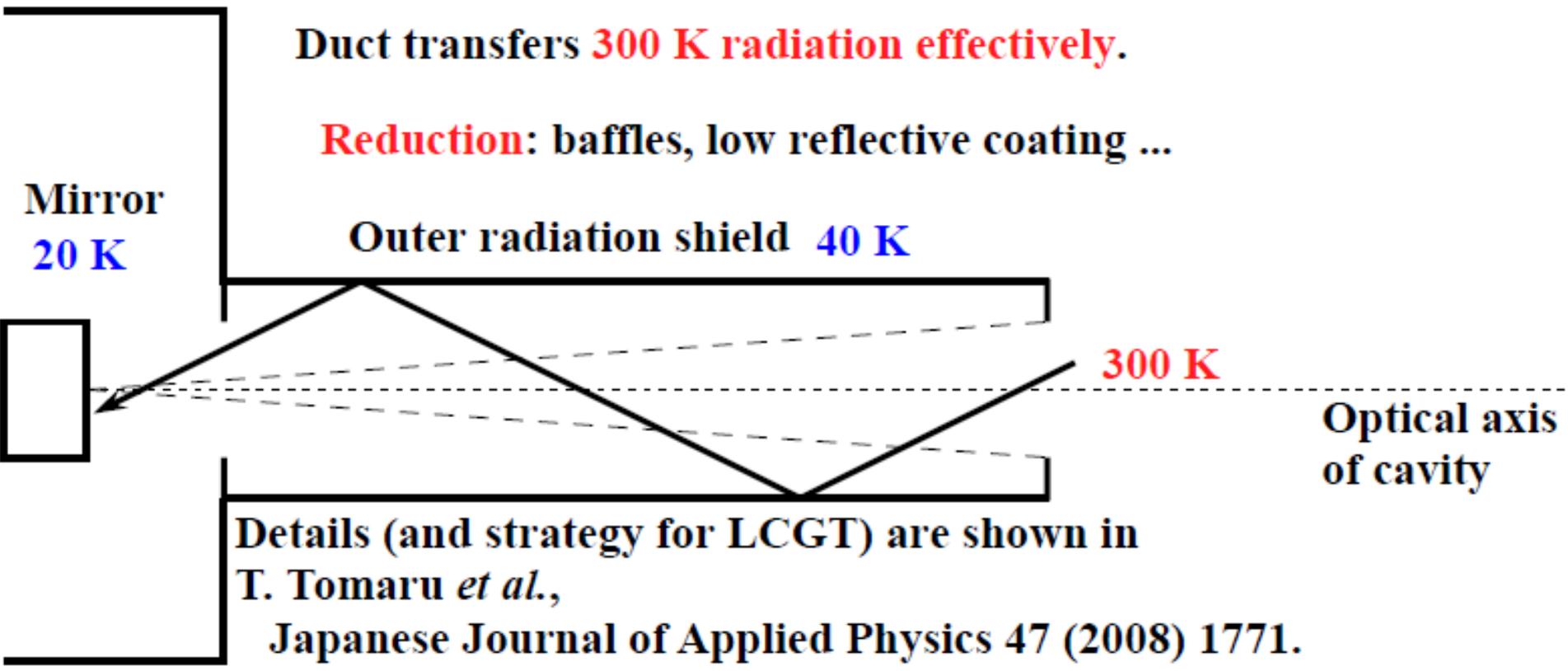


4. How to cool down



Power of 300K radiation is proportional to **solid angle of hole**. So, 300 K radiation should be small. But actually,.....

4. How to cool down



T. Tomaru et al.,
Journal of Physics:Conference Series 122 (2008) 012009.

But actually, cryo duct can transmits 300K radiation well (!) because inner surface of cryo duct reflects radiation !

4. How to cool down

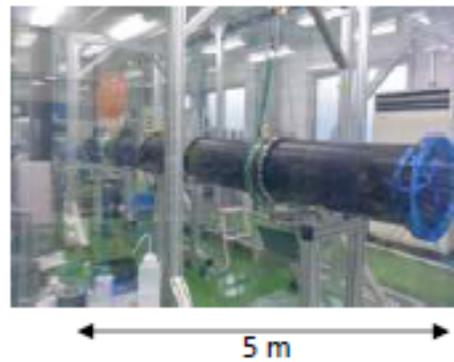
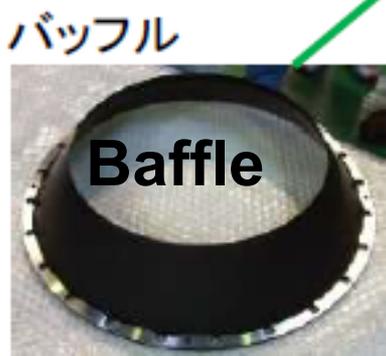
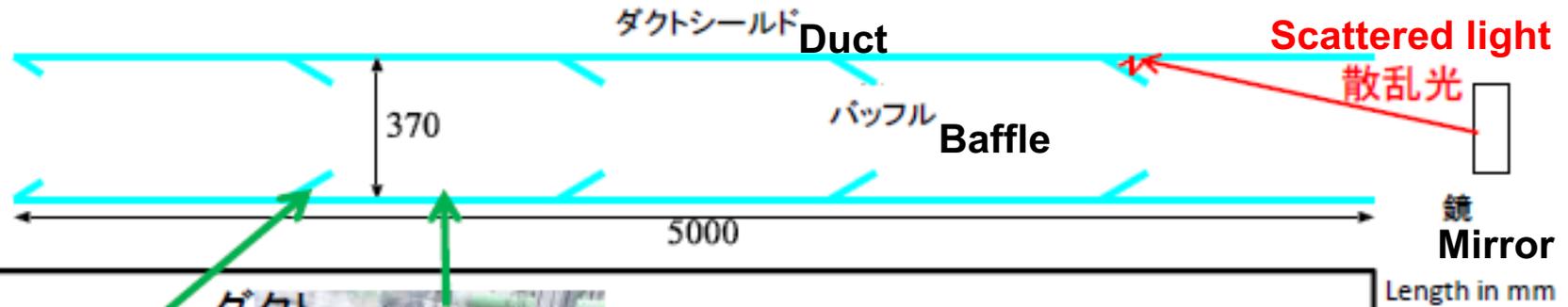
Cryo duct for KAGRA

Y. Sakakibara, Ph.D. thesis (2015).

ダクトシールドの設計

真空チェンバー

Cryostat
クライオスタット



Black coating on inner surface and baffles
Only **0.1 W** radiation can pass through.

4. *How to cool down*

(3) Initial cooling time

It is always **serious issue** in cryogenic experiment.
KAGRA case is introduced.

(a) Initial cooling of **radiation shield**

Otherwise, mirror can not be cooled !

(b) Initial cooling of **mirror and its suspension**

(**cryogenic payload**)

It is isolated thermally !

(c) How to **reduce** cooling time

4. *How to cool down*

(a) Initial cooling of **radiation shield**

In the case of KAGRA

Total mass of inner radiation shield is about **700 kg**.

C. Tokoku *et al.*, CEC/ICMC2013, 2EPoE1-03, Anchorage, USA (2013).

Typical specific heat : **1000 J/Kg/K at 300K**

Dulong-Petit law

4. How to cool down

(a) Initial cooling of **radiation shield**

Heat extraction power

Pulse tube cryocooler : **100 W** (above 60K)

in KAGRA (two cryocoolers).

C. Tokoku *et al.*, CEC/ICMC2013, 2EPoE1-03, Anchorage, USA (2013).

Calculation shows that it takes about **14 days** to cool down (from 300K to 100K).

Below 100K, cooling time is quite small because of small specific heat (and large thermal conductivity in some case).

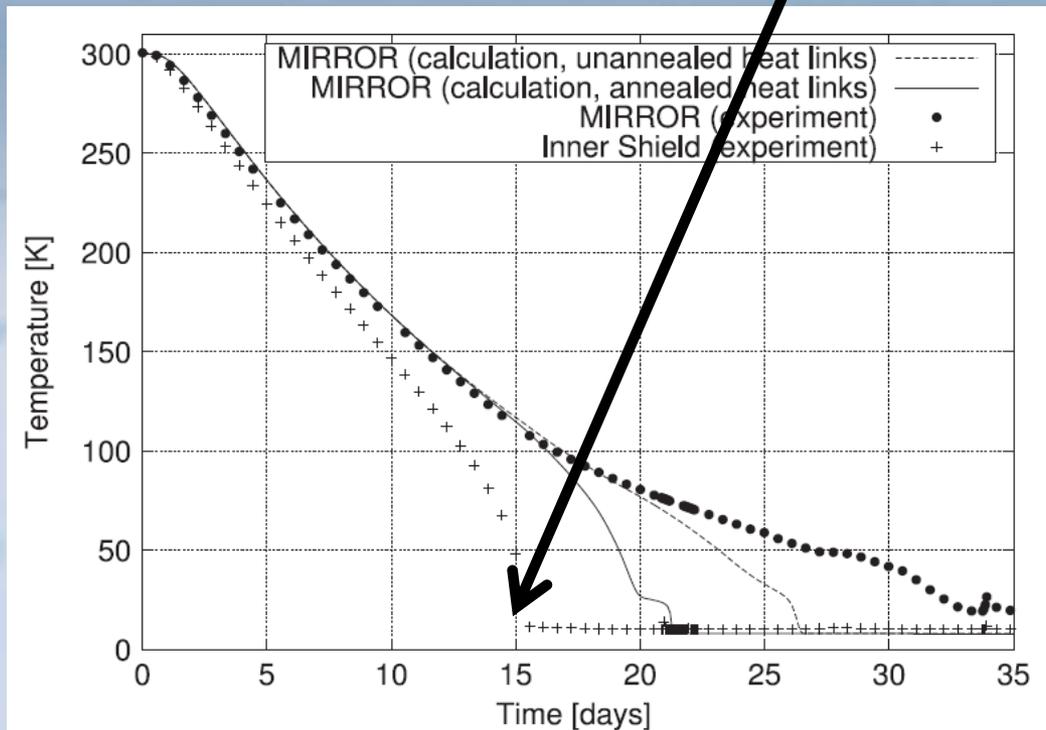
Debye model (Specific heat is proportional to cubic of temperature.)

4. How to cool down

(a) Initial cooling of radiation shield

Experiment showed that it takes **15 days** to cool the radiation shield.

Y. Sakakibara *et al.*, Classical and Quantum Gravity 31(2014)224003.



4. *How to cool down*

(b) Initial cooling of **cryogenic payload**
Payload is **isolated** thermally.

In the case of KAGRA,

Total mass of payload is **200 kg**.

Typical specific heat : **1000 J/Kg/K at 300K**

Dulong-Petit law

4. *How to cool down*

(b) Initial cooling of **cryogenic payload**

Cooling payload **above 100 K**

Black body **radiation**

KAGRA : Total surface area of payload is 1 m^2 .

Radiation power is **460W** at most (**black coating** is necessary) when radiation shields were cooled down.

It takes only **1 day** to cool down cryo payload if radiation shields suddenly cooled !

4. *How to cool down*

(b) Initial cooling of **cryogenic payload**

Cooling payload **above 100 K**

Thus, **temperature of payload is comparable** with that of **radiation shield** while the payload temperature is **above 100K**.

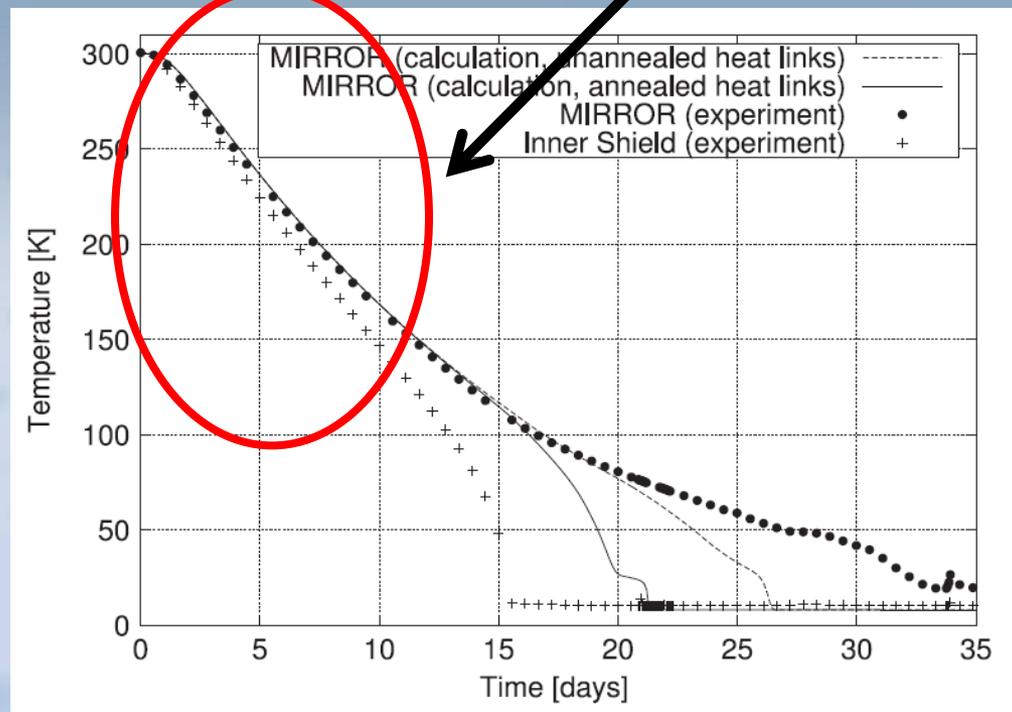
In the case of **Silicon** at **120K**, this is end of story.

4. How to cool down

(b) Initial cooling of **cryogenic payload**

KAGRA : Temperature of payload is comparable with that of radiation shield.

Y. Sakakibara *et al.*, Classical and Quantum Gravity 31(2014)224003.



4. How to cool down

(b) Initial cooling of **cryogenic payload**

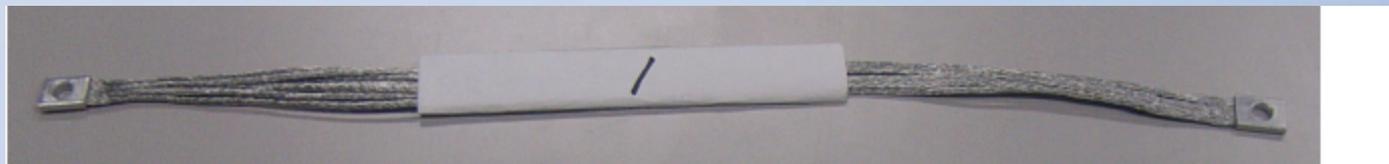
Cooling payload **below 100 K** (in the case of below 20K!)

Radiation does **not work** (T^4).

In the case of KAGRA,

Heat links (twisted thin Al fibers) between payload and cryocoolers.

Thermal conductivity of Al is around 10 K is 10000W/m/K.



アルミ撚り線7本並列 (0.15mm×49本×7列) 30cm, 40cm, 50cm 99

4. *How to cool down*

(b) Initial cooling of **cryogenic payload**

Cooling payload **below 100 K**

Designed heat link can transfer 1W heat power
(mirror heat absorption is **1 W**).

Specific heat is smaller at lower temperature.

Debye model (Specific heat is proportional
to cubic of temperature.)

Typical specific heat : **300 J/Kg/K** at 100K

KAGRA : Total mass of suspension is **200 kg**.

4. How to cool down

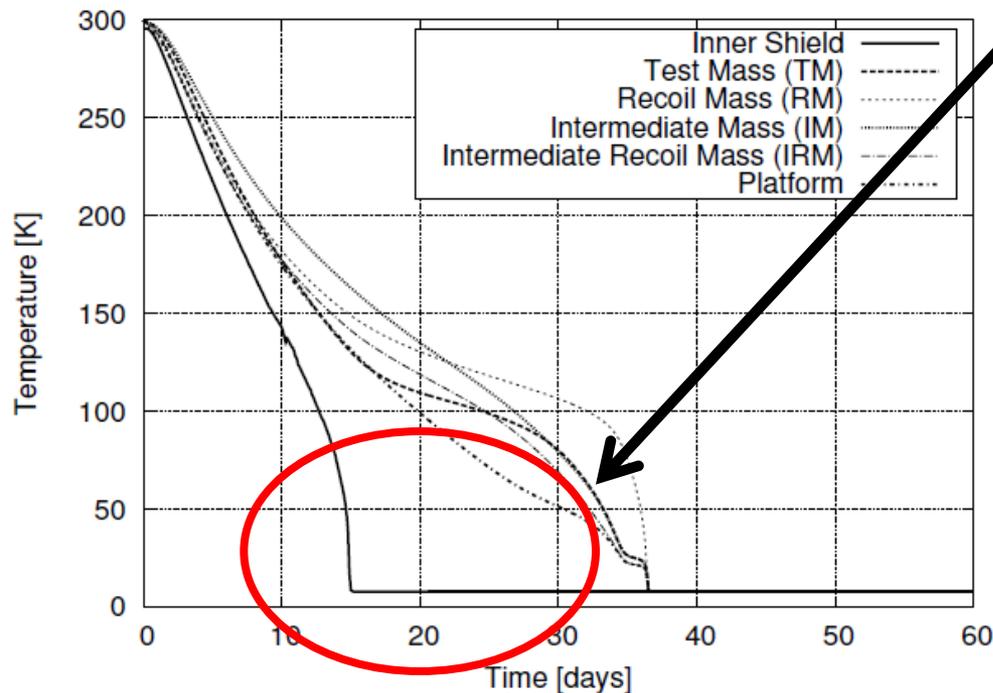
(b) Initial cooling of **cryogenic payload**

Cooling payload **below 100 K**

When power of heat extraction is **1W**,

the cooling time is **15** days.

Y. Sakakibara *et al.*, CEC/ICMC2013, 2EOrD4-03, Anchorage, USA (2013).



4. *How to cool down*

(c) How to reduce cooling time

(i) Short cooling of **radiation shield** in any cases

Payload temperature can follow

shield temperature above 100 K

owing to radiation.

(ii) Short cooling of **payload below 100K**

It is **not necessary** in the case of **120K**.

Y. Sakakibara, Ph.D. thesis (2015).

4. *How to cool down*

(c) How to reduce cooling time

(i) Short cooling of **radiation shield**

Much more cryocoolers ?

(KAGRA : 2 cryocoolers for inner shield
2 cryocoolers for payload)

Geometrical constrain, vibration

More powerful cryocoolers ?

Development, vibration

4. How to cool down

(c) How to reduce cooling time

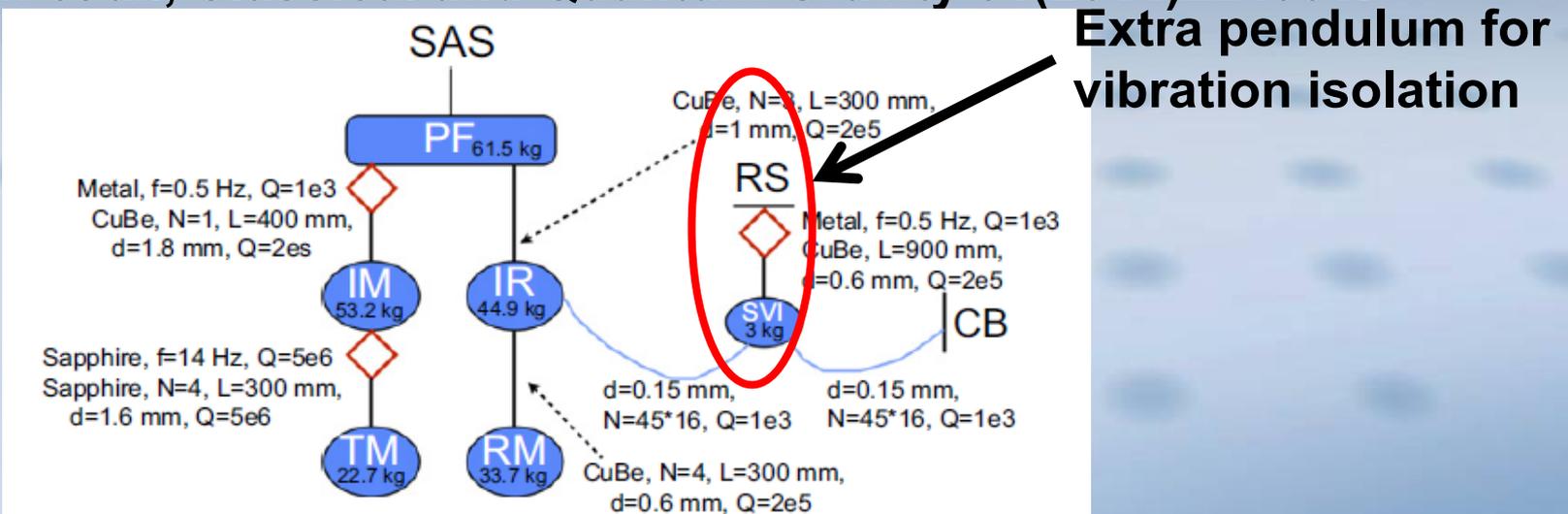
(ii) Short cooling of **payload below 100K**

Thicker and shorter heat links

Issue is **external vibration transmission**.

KAGRA needs extra **vibration isolation system for heat links**.

D. Chen *et al.*, Classical and Quantum Gravity 31(2014)224001.



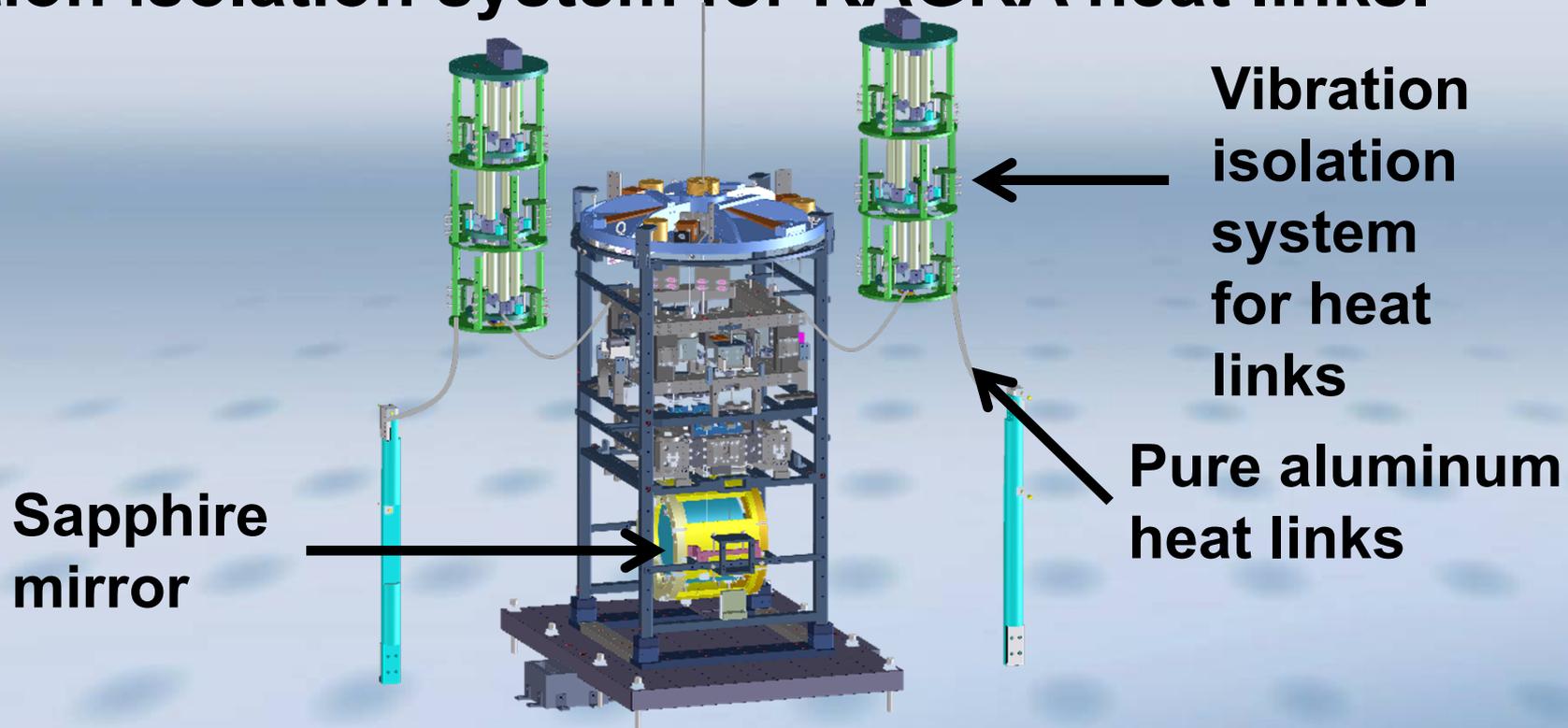
4. How to cool down

(c) How to reduce cooling time

(ii) Short cooling of **payload below 100K**

Thicker and **shorter** heat links

Vibration isolation system for KAGRA heat links.

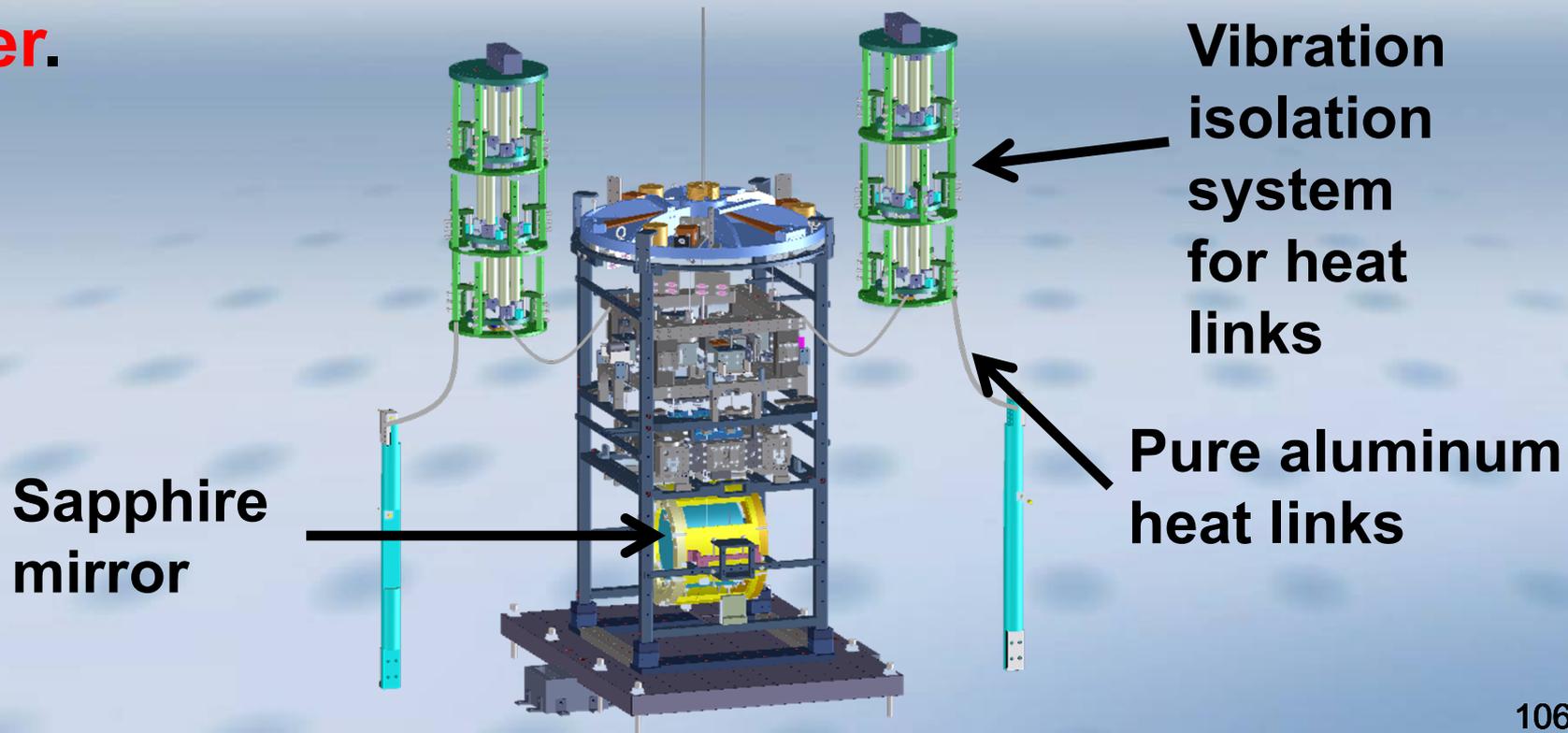


4. How to cool down

(c) How to reduce cooling time

(ii) Short cooling of **payload below 100K**

If you **develop better** heat link vibration isolation system that that of KAGRA, you can adopt thicker and shorter heat links and **initial cooling times is shorter.**



4. *How to cool down*

(c) How to reduce cooling time

(ii) Short cooling of **payload below 100K**

Heat path with **thermal switch**

Initial cooling : Thermal switch is turned **on**. Heat path **works**.

After cooling down : Thermal switch is turned **off**. Heat path is **cut**.

Thermal switch is quite tricky part

4. How to cool down

(c) How to reduce cooling time

(2) Short cooling of **payload below 100K**

Heat path with **thermal switch**

Gas : Radiation **shield should keep gas.**

All holes for laser beam and bar

from room temperature part are closed.

During cooling, main laser beam

can not observe drift of mirror.

Super insulator absorbs helium gas.

It takes longer time to evacuate helium gas.

4. How to cool down

(c) How to reduce cooling time

(2) Short cooling of **payload below 100K**

Heat path with **thermal switch**

Mechanical thermal switch

between payload and crycoolers

Large heat **contact during** initial cooling (force)

Large thermal **isolation after** initial cooling

(soft metal for contact disturbs detachment)

Thermal switch on **suspended object**

Direction and position of mirror should be changed after this switch is turned off.

Mechanism must work at cryogenic temperature.

5. Summary

Merits of **cryogenic** interferometer (Comparison with 300K interferometer)

- (1) Small mirror (coating) thermal noise
- (2) Small thermal lens
- (3) Less serious parametric instability

Challenges

- (1) New materials (sapphire, silicon) as substrate
- (2) Small vibration cooling bath or cryocoolers
- (3) Cryoduct to terminate 300K radiation
- (4) Shorter cooling time of radiation shield

5. Summary

Operation temperature

Below 20K operation

We can enjoy **merits fully**.

What we must **investigate or overcome**

- (1) Thicker fiber for heat transmission
- (2) Heat link vibration isolation
- (3) Cooling time below 100K

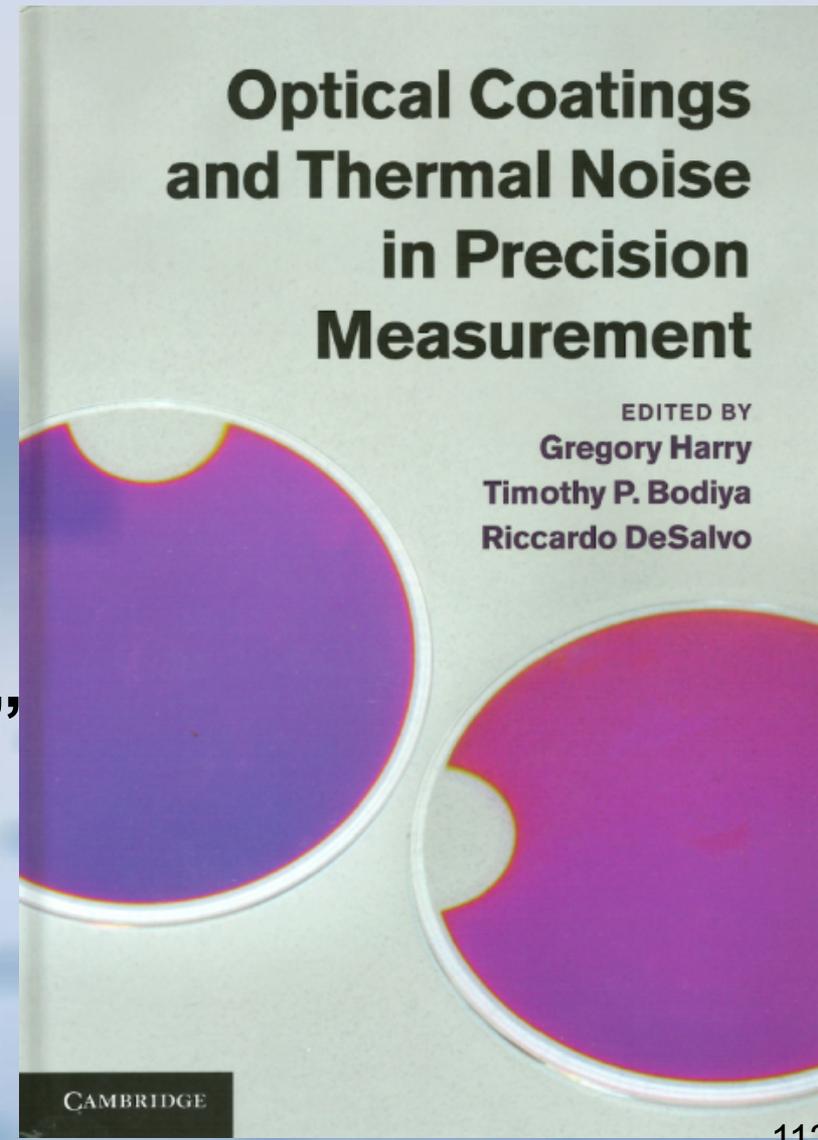
120K operation

Since radiation cooling is **effective**, issues below 20K does **not matter**. But some **merits are missed** ... (Coating mechanical loss reduction is necessary).

The **book about coating thermal noise** was **published** on 2012 !

Cambridge University Press

Chapter 8 (K. Numata and K. Yamamoto) “Cryognics”



Thank you for your attention !

1. Introduction

	aLIGO / AdV	A+/V+	KAGRA	CE 1	CE 2	ET-LF	ET-HF
Arm Length [km]	4 / 3	4	3	40	40	10	10
Mirror Mass [kg]	40 / 42	40	23	320	320	211	200
Mirror Material	silica	silica	sapphire	silica	silicon	silicon	silica
Mirror Temp [K]	295	295	20	295	123	10	290
Suspension Fiber	0.6m/0.7m SiO ₂	0.6m SiO ₂	0.35m Al ₂ O ₃	1.2m SiO ₂	1.2m Si	2m Si	0.6m SiO ₂
Fiber Type	Fiber	Fiber	Fiber	Fiber	Ribbon	Fiber	Fiber
Input Power [W]	125	125	70	150	220	3	500
Arm Power [kW]	710 / 700	750	350	1400	2000	18	3000
Wavelength [nm]	1064	1064	1064	1064	1550	1550	1064
NN Suppression	1	1	1	10	10	1	1
Beam Size [cm]	(5.5/6.2) / 6	5.5/6.2	3.5/3.5	12/12	14/14	9/9	12/12
SQZ Factor [dB]	0	6	foreseen	10	10	10	10
F. C. Length [m]	none	300	unknown	4000	4000	10000	500

T. Uchiyama *et al.*, Physics Letters A 261(1999)5.

2. Thermal noise

In the case of KAGRA (Sapphire, 20K)

